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Neutron Physics Division

TWO BERYLLIUM-MODERATED CRITICAL ASSEMBLIES

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#### ABSTRACT

Two unreflected critical assemblies using beryllium as the moderator and 93.4% enriched uranium metal as the fuel were built to provide a basis for the evaluation of certain reactor calculational procedures. Control and safety rods of the core-element-removal type were used in order that the final assemblies would not be complicated by strong absorber rods. In the first assembly (CA-1), which had outside dimensions of 21.0 x 21.0 x 23.3 in., the 0.01-in.-thick uranium disks were separated by 1-in.-thick blocks of beryllium, which gave a Be: $U^{235}$  atomic ratio of 390 and a fuel loading of 18.08 kg of  $U^{235}$ . The extrapolated value of keff for the system was 1.0054. In the second assembly (CA-18), which had outside dimensions of 24.0 x 28.4 x 24.1 in., the fuel disks were separated by 4-in.-thick blocks of beryllium, which gave a Be: $U^{235}$  ratio of 1560 and a fuel loading of  $U^{235}$ . For this assembly the extrapolated keff value was 1.0020. The observed uranium cadmium fractions in the two assemblies were 0.46 and 0.86, respectively.

A number of multigroup calculations were made to evaluate the effects of various corrections and assumptions. It was concluded that the calculated neutron multiplication is very sensitive to the competition between leakage and slowing down at high energies, a range where fundamental data are uncertain. Without resolving the detailed neutron behaviors in this range, a reasonable selection of data within experimental uncertainties will give satisfactory values for such quantities as critical size.

#### PREFACE AND ACKNOWLEDGEMENTS

During the late 1940's both the Oak Ridge National Laboratory and the Nuclear Energy Propulsion Aircraft Division of the Fairchild Engine and Airplane Corporation in Oak Ridge became interested in performing critical experiments which could be correlated with reactor calculations. Accordingly, a critical experiments facility was constructed and equipped, and plans were made for the two groups to jointly build and operate a simple unreflected beryllium-moderated assembly. The assembly, designated as CA-1, became critical in the early part of 1951. Shortly after the inauguration of the program the NEPA project was terminated and its personnel dispersed, leaving the experimental program incomplete. The Critical Experiments Laboratory continued as an ORNL facility and, after a brief interruption, the experimental program with CA-1 was resumed. Although no complete report of the experiments was ever prepared, some of the results were publicized through private communications, and the discrepancy between the predicted and actual critical size of the assembly gave rise to rather widespread interest. Several attempts, both theoretical and experimental, were made to determine the source of the disagreement, and an interest on the part of ORNL staff members in resolving this problem as well as other inconsistencies associated with the calibration of CA-1 led to the construction in 1954 of a second beryllium-moderated critical assembly, CA-18. Additional calculations were then performed by the author for both assemblies. This report describes the two experiments and presents the calculations. It is to be pointed out that these calculations were completed in 1954 using the best cross-section data available at that time, and the effect of using more recent values has not been investigated.

The success of the two experimental programs is the result of the cooperative efforts of many persons. The equipment and most of the material necessary for the first assembly (CA-1) were procured by the NEPA Group, and the experiment itself was performed with their assistance. The members of this group were:

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#### INTRODUCTION

Multigroup reactor calculation methods were first used in Oak Ridge in 1950 by the NEPA and ORNL Physics Groups, at which time the critical size of an unreflected beryllium-moderated reactor was calculated.<sup>1</sup> In order to experimentally check the validity of the calculational method, a critical assembly was built in the NEPA-ORNL Critical Experiments Facility.<sup>2</sup> This assembly, designated as CA-1, was of the simplest possible geometry, approximating a bare cube. It had a regular lattice of 10-mil-thick enriched uranium metal fuel disks separated by 1-in.-thick blocks of beryllium metal. Besides the ease of construction, such an assembly had the advantage of permitting relatively simple corrections for fuel heterogeneity, and the more complicated calculations necessary for a reflected system could be avoided.

CA-1 was first made critical February 4, 1951. The size of the assembly was roughly 27% less than had been predicted.<sup>1,3</sup> Therefore considerable experimental effort was directed toward finding possible experimental errors. Subsequently a number of calculational attempts were made<sup>4-6</sup> which resulted in more or less rationalizing the discrepancy between theory and experiment. In the spring of 1951 the NEPA project was terminated, resulting in the early curtailment of experiments on this critical assembly.

Until April, 1954, CA-1 remained the only simple beryllium-moderated critical assembly available for analysis. At that time a second assembly, designated as CA-18, was built similar to CA-1 except that the fuel spacing was changed from 1 to 4 in. It was recognized that an assembly having a higher uranium concentration than that of CA-1 would have been more desirable from the analytical standpoint, since the greatest uncertainties in cross-section data are in the high-energy range. However, the 4-in. spacing was chosen on the basis of ease of construction and the availability of materials and time.

Upon completion of CA-18 several multigroup calculations were performed and compared with the actual critical assemblies. These calculations for CA-1 were also compared with the earlier calculations. The purpose of this report is to describe the two critical assemblies in detail, to discuss the experimental measurements which were made in connection with them, and to evaluate various assumptions made in the multigroup calculations.\*

<sup>1.</sup> A. O. Mooneyham, NEPA-1710 (1951)

<sup>2.</sup> F. T. Bly et al., "NEPA Critical Experiment Facility," NEPA-1769 (1951).

<sup>3.</sup> A. O. Mooneyham, IC-51-2-7 (Supplement to NEPA-1710) (1951)

<sup>4.</sup> G. M. Safanov, YF-10-45 (1951)

<sup>5.</sup> D. K. Holmes, ORNL-1227, p. 64

<sup>6.</sup> C. B. Mills, ORNL-1493 (1953)

<sup>\*</sup> Calculations reported here were completed in 1954 using the best data available at that time. The effects of subsequent improvements in data have not been thoroughly investigated. However, the experimental results are still valid and will continue to serve as a comparison for subsequent calculation techniques.

#### I. DESCRIPTION OF THE ASSEMBLIES

The matrix into which the materials for both of the assemblies were placed was a 6-ft cube consisting of 3-in.-OD square 2S aluminum tubes, each having a wall thickness of 0.047 in. This matrix, which is shown in Fig. 1 for CA-1, was divided vertically into halves which could be separated by remote control. Fuel elements were placed in both halves and the assembly was completed by bringing them together.

Core elements consisted of 93.4% enriched, 2.860-in.-dia, 0.01-in.-thick uranium disks (average mass of  $U^{235} = 16.774$  g) separated by one (CA-1) or four (CA-18) 1-in.-thick by 2-7/8 in. by 2-7/8 in. beryllium metal blocks of average density 1.86 g/cc. These were held together by 3/16-in.-dia aluminum skewers through 0.196-in.-dia center holes in the fuel disks and the beryllium blocks. The elements were designed to maintain a constant fuel spacing throughout the assembly.

The arrangements of the core elements in the aluminum grid for CA-1 and CA-18 are indicated in Figs. 2 and 3, respectively. Row 8 of CA-18 contained quarter-sized core elements. They were 1-7/16 in. square but otherwise had the same composition as the other elements.

Control and safety rods were similar to other core elements except that provision was made for remotely moving them within the assembly. The use of this type of control and safety devices permitted an assembly free of perturbing neutron absorbers. In order to give these elements more structural strength, stainless steel rather than aluminum skewers were used. The safety rods were inserted by compressed air, spring loaded and magnet held. All safety rod elements were inserted or withdrawn by means of a screw drive mechanism. One such element is shown displaced from its normal position in Fig. 1.

For CA-18 a simple servo control system was devised<sup>7</sup> which employed a 1X Brown "Electronic"\* amplifier and a 27-rpm balancing motor arranged to insert or withdraw one of the core elements. This control element was located in cell K-13, as shown in Fig. 3. The input signal was derived from one of the neutron monitoring instruments in opposition to an adjustable demand potential supplied by a 1.5-v battery. The system was capable of changing the reactivity of the assembly at a maximum rate of 0.01%/sec, which was adequate to follow slow transients or to maintain stable operations.

A comparison of the physical compositions of the two assemblies is given in Table 1. For completeness the corresponding  $k_{eff}$  values are included. These values correspond to the regular assembly having the indicated fuel loadings extrapolated to the condition of all control rods fully inserted. Criticality in both assemblies was reached with one control rod slightly withdrawn. Each stainless steel volume fraction includes the stainless steel

<sup>7.</sup> M. E. Remley, Science 119, 29 (1954).

<sup>\*</sup> Minneapolis-Honeywell Regulator Company, Philadelphia 44, Pa.



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1, 2, 3, 4 = SAFETY RODS IN MOVABLE HALF 5, 6, 7, 8 = SAFETY RODS IN FIXED HALF A, B = CONTROL RODS IN MOVABLE HALF C, D = CONTROL RODS IN FIXED HALF

FUEL ELEMENT DETAIL



Fig. 2. Arrangement of Core Elements in CA-1 (Fixed Half Interface)

5



FUEL ELEMENT DETAIL



Fig. 3. Arrangement of Core Elements in CA-18 (Fixed Half Interface)

	CA-1	CA-18
Outside Dimensions 2	1.0 x 21.0 x 23.3 in.	24.0 x 28.4 x 24.1 in.
Spacing of 10-mil-thick fuel disks	l in.	4 in.
Volume fractions:		
Aluminum (grid and skewer	s) 0.0610	0.0611
Beryllium (moderator)	0.9060	0.9128
Uranium (fuel)	0.0064	0.0016
Stainless steel (skewers)	0.0004	0.0003
Void.	0.0262	0.0242
U <sup>235</sup> loading	18.08 kg	7.65 kg
k <sub>eff</sub>	1.0054	1.0020
Be:U <sup>235</sup> atomic ratio	390	1560

skewers plus an artificial quantity having the same absorption cross section as the impurities in other parts of the system. Table 1 and Figs. 2 and 3 describe the nearest approaches to simple unreflected cubical critical assemblies for each of the two fuel spacings. In subsequent experiments the necessary excess multiplication was provided by placing additional core elements just outside of the clean assembly.

#### II. EXPERIMENTAL MEASUREMENTS

#### Measurements with CA-1

Initial Approach to Criticality. In the approach to criticality with CA-1 (see Fig. 2) uranium fuel disks were added to the core elements nearest the center of a  $24 \times 24 \times 25$  in. beryllium-reflected array until the assembly became critical. Except for one missing corner fuel element, this first array consisted of a  $4 \times 4$  cell core with a 6-in.-thick reflector on four sides. It contained 6.05 kg of U<sup>235</sup>. At criticality one control rod in the reflector (rod D in cell M-10) was withdrawn 3.7 in. The unreflected critical assembly was reached by alternately removing outside beryllium and adding fuel. The final unreflected parallelepiped had the dimensions 21.0 x 21.0 x 23.22 in., each of the 49 elements consisting of a 23-in. total thickness of beryllium and 22 fuel disks of average thickness 0.01 in. The assembly contained 18.08 kg of U<sup>235</sup> and was critical with control rod C withdrawn 2.59 in., corresponding to 0.0054 in reactivity.

The experimental critical mass of the unreflected assembly was approximately 27% below that predicted in the multigroup calculations.<sup>8</sup> Since this disagreement was larger than expected, the experimental work described below was performed in an effort to locate any possible experimental errors of this magnitude.

Control Rod Calibration. In a control rod calibration, rod C was found to have a total value of \$2.82 by comparison with rod-drop measurements of the symmetrically located safety rods 7 and 8. The calibration curve shown in Fig. 4 was constructed by assuming a form similar to the detailed calibration measured in CA-18 and normalizing to a total value of \$2.82. This curve was used for subsequent reactivity comparisons. No detailed control rod calibration measurements were made on CA-1.

<u>Reactivity Measurements</u>. The experiments intended to determine the effect on the reactivity of spurious reflections from surrounding materials are summarized in Table 2. Reactivity values were found by comparing known control rod positions with the calibration of Fig. 4. In the first experiment listed, the neutron trap below the assembly was designed to cut off any reflected neutrons from the floor and structure; however, the reflected neutrons from the trap were found to be more effective than those from the floor. In one experiment (No. 7) the entire parallelepiped was moved up 27 in. to the

8. A. O. Mooneyham, NEPA-1710, op. cit.





Fig. 4. Calibration of Control Rod C in CA-1.

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Experiment Number	Description of Material <sup>a</sup>	Gain in Reactivity Over Basic Assembly (¢)
l	36 x 45 in. neutron trap was added below assembly; trap consisted of 8 in. of air space, 0.017 in. of cadmium, 1 in. of plexiglas, and 3 in. of graphite <sup>b</sup>	31.5
2	Same as No. 1 except a 1/4-inthick by 36 in. boral sheet was added 2-1/2 in.below the assembly (i.e., between the assembly and the cadmium)	19.8
3	Same as No. 2 except all plexiglas and graphit was removed	e 3.5
ц	1/4-inthick boral sheet was placed on four sides of assembly (no air spaces)	73.5
5	l-inthick boral sheet was placed 2 in. from bottom of assembly (i.e., a 2-inthick air ga existed between the assembly and the boral)	p 37.3
6	Layer of pressure tape covered one side of assembly (similar tape was used to cover the boral plates used in the above measurements)	4.5
7	Entire critical assembly was raised 27 in. to uppermost position in the aluminum grid	69.0
8	Two pieces of 3 in. by $1-3/8$ in. steel channel were added to the top of the assembly; the assembly remained in the uppermost position in the grid	e 88.6
9	Assembly was returned to the center of the gri and a 1-inthick aluminum sheet was added to the top and one side of the assembly	d, 126.9
10	One of the 1-inthick aluminum sheets used in No. 9 was moved out, leaving a 3-inthick air gap between the assembly and the aluminum	26.8

#### Table 2. Effect of Various Surrounding Materials on the Reactivity of CA-1

a. The aluminum grid was supported on a 3/4-in.-thick steel table top, 26 in. above the floor.

b. Listed in order of placement from bottom of assembly.

c. Four similar channels were part of the normal structure.

d. The average total thickness of aluminum in the grid outside of the assembly was 1.57 in.

uppermost position in the grid. Again a gain in reactivity resulted, this time caused by an increase in the number of neutrons reflected from the structural steel used to hold the aluminum grid in place. Experiments 9 and 10 were performed in order to exaggerate the effects of the external aluminum grid and thereby set an upper limit on this contribution.

Some inconsistencies in the control rod positions for criticality were observed in the above measurements. All movable parts of the system were found to be mechanically reproducible, and the reason for the inconsistencies had not been ascertained at the time the assembly was dismantled. The difficulty was later attributed to photoneutrons from the beryllium, but no quantitative conclusions could be drawn.

<u>Miscellaneous</u> Measurements. One measurement of the uranium cadmium fraction (CF)\* was made by comparing the activation of the fuel disk nearest the geometric center of the assembly to that of a similar fuel disk covered with 0.02 in. of cadmium. This gave a cadmium fraction value of 0.46. The accuracy of this measurement is questionable owing to uncertainties in the fuel disk background as well as in the counting correction factors.

Power distributions within the assembly were observed by measuring the gamma-ray activity of fuel disks. The results were in substantial agreement with the expected cosine distribution as indicated in Figs. 5 and 6. The terms lateral and longitudinal are used to define directions parallel or normal, respectively, to the interface of the two assembly halves.

A danger coefficient measurement on the fuel was made by removing the fuel disk nearest the center of the assembly and replacing it with a smaller fuel disk. The observed loss in specific reactivity was  $2.9 \not{e}/g$  of  $U^{235}$ . Assuming the fuel importance to be proportional to the flux squared and assuming the value  $100\not{e}$  to be equivalent to 0.0073 of the reactivity, it was found that  $\delta k/k = 0.475 \ Gm/M$ , where M is the mass of  $U^{235}$  in the critical assembly.

#### Measurements with CA-18

Initial Approach to Criticality. The first critical array of CA-18 (Fig. 2) had an over-all size of 27 x 27 x 24 in. Except for the four missing corner fuel elements, the array consisted of a 7 x 7 cell core with a 3-in.-thick beryllium reflector on four sides. It contained 4.97 kg of  $U^{235}$  and was critical with one control rod in the reflector (rod D) displaced 2.25 in. The final unreflected critical assembly had dimensions 24.00 x 28.40 x 24.06 in. and contained 7.65 kg of  $U^{235}$ . This unreflected assembly was critical with control rod A in the core displaced 2.60 in., corresponding to 0.0020 in reactivity.

<u>Control Rod Calibration</u>. An absolute calibration, that is, one that is independent of delayed neutrons or transients, was made for CA-18. This was done by observing the displacement of control rod A caused by the addition of

<sup>\*</sup>  $CF = (N_b - N_{Cd})/N_b$ , where  $N_b$  is the activation of the bare disk and  $N_{Cd}$  is the activation of the disk having a 20-mil-thick cadmium cover on each side.







ORNL-LR-DWG 14833



Fig. 6. Power Distribution Along Several Longitudinal Traverses in CA-1 (Gamma-Ray Activity of Uranium Disks).

a 1/4-in.-thick layer of beryllium on top of the assembly and assigning to the displaced portion of the rod the same reactivity value as that calculated for a change in the buckling of the system caused by the addition of the beryllium. The calculation, which is presented in Appendix A, is probably accurate within 2%. Control rods B, C, and D were then calibrated against control rod A.

The control rod sensitivity, i.e., the change in reactivity per inch of travel, is shown in Fig. 7 for each of the rods in CA-18. The control rod position is the displacement from the center of the assembly, or the distance the rod is withdrawn. The maxima particularly prominent in the sensitivity curve for rod A occur when the fuel planes in the rod are misaligned with those in the stationary part of the assembly. The steep part of the curve in the vicinity of the edge of the assembly is due to the effect of "plugging" the control rod channel. This effect may be otherwise thought of as due to adding moderator in its region of highest importance. The control rod calibration curves in Fig. 8 were found by integrating the sensitivity curves from infinity to any particular position.

Conventional rod drop and pile period observations were also made for comparison with the absolute calibration of the rods; however, these calibration methods depend on a knowledge of the effectiveness of delayed neutrons and are complicated by the presence of photoneutrons from beryllium. Pile period observations are particularly difficult owing to a number of photoneutron groups arising from long-lived fission-fragment hard gamma-ray emitters. Table 3 is a comparison between the reactivity values determined from the pile period observations and those obtained in the absolute calibrations. Reactivity values from the pile period observations were calculated from five delayed neutron groups? through use of the in-hour equation, neglecting delayed photoneutrons from the beryllium. The inconsistency of the ratio of the results from the two methods reflects the inherent uncertainty in the effective value of delayed neutrons, and the effect of the delayed photoneutrons from the beryllium. Further, the apparent period is strongly influenced by the operational history of the system.

In Table 4 the control rod reactivity values obtained by the rod drop method, using a total delayed neutron fraction of 0.0073, are compared with the total control rod reactivity values obtained by the absolute calibration method (see Fig. 8). Again, the lack of consistency is no worse than expected in view of the experimental uncertainties in the two methods.

Both the pile period and the rod drop results are about 80% of the corresponding absolute reactivity value. Apparently the best values found by the pile period method are those from the shorter period range.

Foil Exposures. Several types of foil exposures were made within CA-18. Indium-aluminum foils were used to obtain indium activation measurements. In addition, fission rates within the assembly were determined by observing the activity of fission fragments on thin aluminum "catcher" foils which were in direct contact with fuel disks during an exposure. The activities of similar

<sup>9.</sup> S. Glasstone and M. C. Edlund, "The Elements of Nuclear Reactor Theory," p. 65, D. Van Nostrand Company, Inc., 1952.



Fig. 7. Change in Reactivity of CA-18 as a Function of Withdrawal of Control Rods.



Fig. B. Calibration Curves for Control Rods in CA-18.

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		Reactiv	vity	Ratio of
Observation	Period (sec)	Pile Period Method	Absolute Method	Reactivities, Pile Period:Absolute
1	770.0	0.000117	0.000182	0.646
2	521.0	0.000170	0.000460	0.369
3	417.0	0.000209	0.000347	0.603
ŭ	166.6	0.000471	0.000610	0.773
5	152.0	0.000507	0.000588	0.863
6	101.3	0.000697	0.000873	0.799
7	95.6	0.000733	0.000920	0.798
ė	95.6	0.000733	0.00111	0.661
9	84.8	0.000795	0.000940	0.846

Table 3.	Comparison of Reactivity Values of Control Rods in
-	CA-18 Determined by Pile Period and by Absolute
	Calibration Methods

Table 4. Comparison of Reactivity Values of Control Rods in CA-18 Determined by Rod Drop<sup>2</sup> and Absolute Calibration Methods

	Equivalent	•	•	Rea	ctivity	Ratio of	
Safety Rod	Control Rod	No	Ni	Rod Drop Method	Absolute Method	Reactivities, Rod Drop:Absolute	
1	C .	39.0	35.0	0.000831	0.00143	0.581	
- 3	D	34.8	19.1	0.00600	0.00793	0.757	
6	В	38.0	18.0	0.00810	0.01070	0.757	
5	А	37.5	9.1	0.02385	0.02564	0.928	

.

a. In the rod drop method,  $\delta k/k = 0.0073 (N_0 - N_1)/N_1$ , where  $N_0$  is the initial power level and  $N_1$  is the power level extrapolated to the time immediately following the rod drop; see, for example, Ref. 9, p. 305, Eq. 10.35.3.

catcher foils which were exposed within a composite fuel disk, that is, a disk consisting of five individual disks, were used to study the effect of the self shielding of the fuel.

The indium-aluminum foils were each 5/16 in. in diameter and 10 mils thick, having an effective indium thickness of 0.3 mil or  $5.4 \text{ mg/cm}^2$ . For one exposure the foils were placed along a horizontal lateral line (that is, parallel to the fuel plane) which was 1/4 in. from the midplane. Plots of the relative activity of the foils for this traverse are shown in Fig. 9 and are in substantial agreement with the expected cosine distribution. Similar measurements (Fig. 10) along the longitudinal axis of the assembly (that is, perpendicular to the fuel plane) indicate sharp depressions in the flux near the fuel planes, but the plots are enveloped by cosine curves. The indium-cadmium fraction - - obtained by comparing the activity of bare foils to that of foils covered with 20-milthick cadmium covers - - varies from a maximum of 0.4 between the fuel planes to a minimum of 0.3 at the fuel planes.

Fission rates determined by the catcher foil technique are shown in Figs. 11 and 12 for foil exposures along the lateral and longitudinal axes, respectively. In some cases measurements on opposite sides of the same fuel disks are indicated. The cadmium fraction for fuel is seen to vary between 0.852 and 0.880. (The cadmium covers used in these measurements were also 20 mils thick.)

The variation of the fission rate within the composite fuel disk consisting of five 2-mil-thick disks is shown in Fig. 13. This composite disk replaced one of the conventional 10-mil-thick disks during the measurements. The average fission rate throughout the 10 mils was found to be 89.2% of the average surface fission rate on the disk.

<u>Reactivity Measurements</u>. A number of experiments were performed to determine the effect on the reactivity of CA-18 of introducing various materials into the assembly. In the first group of experiments, the core element in cell K-13 (see Fig. 3), the cell nearest the center of the assembly, was pushed back from the midplane so that a sample of material could be placed very near the center of the assembly. The changes in reactivity caused by various materials in this position are summarized in Table 5. All of the samples were 2-7/8 in. square having the thicknesses indicated in Table 5. The reactivity values listed are found from the changes in control rod positions necessary to compensate for the addition of the sample into this space, that is, the difference in the reactivity of the assembly with the sample in place and the reactivity of the assembly with an air space of equal thickness at the same location.

The effects on the reactivity of other materials were observed in a similar way in cell L-15. The results of these tests are listed in Table 6. The liquid samples were contained in a thin-walled stainless steel can,  $1 \ge 2-7/8 \ge 2-7/8$  in., and their reactivity values were corrected for the effect of the can. Furfural, H4C502, is an organic material having a hydrogen density approximately one-half that of water. The sample labeled "1/2 Plexiglas" was prepared by drilling small holes in a Plexiglas block to reduce its mass from 157.7 g to

1.2 0 1.0 DASHED LINES ARE CALCULATED COSINES σ O BARE INDIUM FOILS △ CADMIUM-COVERED INDIUM FOILS 0.9 D BARE MINUS CADMIUM-COVERED INDIUM FOILS 0 CADMIUM FRACTION FROM THE BEST FIT COSINES IS 0.364 0.8 Δ EDGE EDGE OF ASSEMBLY n 0.3 EXTRAPOLATED П Δ 0.2 П 0 0.1 0 2 3 4 5 6 7 0 8 9 10 11 1 12 13 14 15 16 DISTANCE FROM CENTER PLANE (in.)

Fig. 9. Activation of Indium Foils Along a Horizontal Lateral Traverse (¼ in. from Midplane) in CA-18.

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Fig. 10. Activation of Indium Foils Along the Longitudinal Axis of CA-18.

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Fig. 11. Power Distribution Along a Lateral Axis of CA-18 (Fission Rate of Fuel Disks).

# DASHED LINE IS CALCULATED COSINE • BARE FUEL DISK; CATCHER ON SIDE NEAREST CENTER PLANE O BARE FUEL DISK; CATCHER ON SIDE AWAY FROM CENTER PLANE ▲ CADMIUM-COVERED FUEL DISK; CATCHER ON SIDE NEAREST CENTER PLANE $\Delta$ CADMIUM-COVERED FUEL DISK; CATCHER ON SIDE AWAY FROM CENTER PLANE -■ CADMIUM FRACTION; CATCHER ON SIDE NEAREST CENTER PLANE □ CADMIUM FRACTION; CATCHER ON SIDE AWAY FROM CENTER PLANE



1.2

Fig.12. Power Distribution Along the Longitudinal Axis of CA-18 (Fission Rate of Fuel Disks).

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Fig. 13. Self-Shielding Factor for a 10-mil-Thick Fuel Disk in CA-18.

Sample	Thickness (in.)	Mess (g)	Atomic Weight (g/mole)	Number of Moles	Reactivity, <sup>a</sup> Sk±0.00002	らk/mole (x10 <sup>-3</sup> )
Mg	0.500	115.78	24.32	4.760	+0.00024	+0.0504
Fe	0.500	523.68	55.85	9.376	-0.00902	-0.963
Ti	0.320	185.67	47.90	3.876	-0.00936	-2.42
Al	0.500	179.3	26.97	6.648	-0,00864	-1.301
Ni	0.160	183.15	58 <b>.</b> 69	3.121	-0.00584	-1.870
Mo	0.425	475.2	95.95	4 <b>.</b> 953	-0.01050	-2.120
Ср	0.250	277.42	92.91	2.986	-0.00335	-1.121
Teflor	0.500	152.28	-	-	+0.00001	-

Table 5. Reactivity Values of Various Materials Inserted in Cell K-13 Near the Center of CA-18

a. Compared to reactivity of assembly with a void of equal volume.

Table 6. Reactivity Values of Various Materials Inserted in Cell L-15 Near the Center of CA-18

Sample	Thickness (in.)	Mass (g)	Reactivity, <sup>a</sup> Sk±0.00005
Furfural <sup>b</sup>	1.0	150.1	-0.00005
Water <sup>b</sup>	1.0	127.6	-0.00009
1/2 Plexiglas	1.0	83.5	+0.000003
Graphite	7.25	1624.3	+0.00010
Teflon	7.2岁	2026.02	-0.00005

a. Compared to reactivity of assembly with a void of equal volume; values for Furfural and water samples were corrected for stainless steel containers.

b. Contained in thin-walled stainless steel can.

83.5 g. Teflon  $(CF_2)_n$  is of interest as a common material used in other zero power critical assemblies. Since cell L-15 is off center, absorption and scattering effects of the samples somewhat cancel each other. No significant differences between water, Furfural, 1/2 Plexiglas, and Teflon are apparent, although the large graphite block was observed to be significantly better as a scatterer than the Teflon block of the same size.

The effect of the fuel heterogeneity in the assembly was determined by observing the increase in the reactivity caused by replacing the 10-mil-thick disks in one-half of the element in cell L-14 with uniformly spaced 2-milthick disks having the same total mass. In another test the number of 2-milthick disks used was reduced such that the total reactivity value of the element was approximately the same as that of a normal element. The results of these observations are presented in Table 7. For purposes of comparison, the reactivity values of cell M-13 with a normal fuel loading and without any fuel are also given. Cell L-14 is nearer the center of the assembly than cell M-13, which explains the higher reactivity values. The results given in Table 7 are only a qualitative indication of self shielding. Since the neighboring fuel elements were not changed, the effectiveness of the thin disks in positions intermediate to the normal fuel positions was exaggerated.

	Half-	Cell Descriptio	on	Re	activity Valu	1e
Cell No.	Number of Fuel Disks	Thickness of Each Disk (mils)	Total U <sup>235</sup> Mass (g)	Of Total Half-Cell	Of U <sup>235</sup> in Half-Cell	Of 1 g of U <sup>235</sup> in Half-Cell
L-14	0	_	0.0	0.02307 <sup>a</sup>	_	-
L-14	3 <sup>b</sup>	10	50.3	0.02890	0.00583	0.000116
L-14	15	2	50.3	0.03222	0.00915	0.000182
L-14	8	2	26.8	0.02873	0.00566	0.000211
M <b>-</b> 13	0	-	0.0	0.02204 <sup>a</sup>	-	-
M-13	3 <sup>b</sup>	10	50.3	0.02760	0.00556	0.0001105

Table 7. Reactivity Value of Fuel in CA-18

a. Represents reactivity value of all materials in half-cell except fuel.

b. Normal loading for half-cell.

#### III. MULTIGROUP CALCULATIONS

Several reactor multigroup calculations were performed and the resulting values of the effective multiplication of CA-1 and CA-18 were compared to the experimental values. The method of calculation and the notation were basically the same as those reported earlier.<sup>10</sup> In order to show the approximations used, the derivation of the criticality equation is repeated in Appendix B.

Eight variations of the basic calculations were used to compare the effects of various corrections. In the paragraphs below the assumptions common to all the calculations are discussed first. These are followed by descriptions of the variations in the calculations, a summary of which is given in Table 8. The results of the calculations are then presented and comparisons are made.

General Calculation Procedure. The calculations were made using 32 energy groups from 107 ev to thermal. The cross-section data were taken from published curves<sup>11</sup> and logarithmic averages were found for each energy group. The crosssection values and the fission source distribution are given in Appendix C.

The uranium macroscopic cross sections in each group were reduced to values corresponding to an equivalent homogeneous system. As discussed below, corrections were made in some of the calculations to account for fuel lumping and the associated self-shielding effect.

The scattering of neutrons in the uranium was considered to be isotropic for all energy groups. Except in the special cases specified below, the scattering in the beryllium was also assumed to be isotropic.

The effective leakage cross section,  $DB^2$ , for each group includes the variation of  $\lambda_{tr}$  with energy since  $D = 1/3 \lambda_{tr}$  and  $B^2$  is a function of the extrapolation distance, 0.71  $\lambda_{tr}$ .

#### Variations in Calculations

<u>Uranium Self-Shielding Corrections.</u> With the exception of Method I, one of two different corrections<sup>12</sup> was made in each calculation to compensate for the lumping of the fuel in the experimental assemblies. One of the corrections, called the P<sub>0</sub> correction, was calculated under the assumption that the neutrons entered the absorber isotropically. The second, i.e., the P<sub>1</sub> correction, allowed for some directional preference or a flux depression outside of the uranium. Methods of calculating both corrections are given in Appendix D. The self-shielding factors resulting from these calculations are compared in Fig. 14.

<sup>10.</sup> C. B. Mills, ORNL-1493, op. cit. pp5, 115-117.

<sup>11.</sup> Publications by the Neutron Cross Section Advisory Group, AECU-2040, BNL-170, BNL-170A, BNL-170B.

<sup>12.</sup> J. C. Bartels, "Self-Absorption of Monoenergetic Neutrons," KAPL-336 (1950).

ethod Number	Code Description
I	A, B-1, C-1, D-1, E-1, F-1, G-1
II IIa	A, B-2, C-2, D-1, E-1, F-1, G-1
III IV	A, B-2, C-1, D-2, E-1, F-1, G-1 A, B-3, C-1, D-2, E-1, F-1, G-1
V	A, B-3, C-1, D-2, E-1, F-1, G-2 A B-3, C-1, D-2, E-2, F-1, G-1
VI VII	A, B-3, C-1, D-2, E-1, F-2, G-1

Table 8. Description of the Various Calculations

Code Definitions:

A. Multigroup calculation using general calculation procedure described on page 25 and in reference 6. B-1. No correction made for uranium self shielding. 11 11 11 n 11 B-2. Po 11 11 11 11 11 11 B-3. P1 C-1. Elastic scattering in U<sup>235</sup> for all energies. C-2. Inelastic """ energies>1 Mev. D-1. Isotropic scattering in Be for all energies. D-2. Anisotropic """ "energies >1 Mev. E-1. Elastic scattering in Be for all energies. E-2. Inelastic """ energies >1 Mev. E-2. Inelastic F-1. Neutron slowing-down density of  $\emptyset \xi \Sigma_t$ . น้ำ 11 11 " $\emptyset \xi \Sigma_s$  (Fermi model). 11 F-2. G-1. No correction for effect of external aluminum grid. G-2. Correction for effect of external aluminum grid.



Fig. 14, Uranium Self-Shielding Factors Calculated by  $P_0$  and  $P_1$  Approximations.

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Elastic Versus Inelastic Scattering in  $U^{235}$ . In general, the neutron scattering in uranium was assumed to be elastic. However, an inelastic scattering correction was included in the IIa calculation of CA-1. Since the effect was small, a similar calculation was not made for CA-18 in which the average neutron energy was lower.

Isotropic Versus Anisotropic Scattering in Beryllium. In methods I, II, and IIa, the neutron scattering in beryllium was considered to be entirely isotropic. In methods III through VII corrections were made for anisotropic scattering of neutrons with energies higher than 1 Mev. These corrections affected the multigroup calculation through variations in the transport cross section,  $\sigma_{tr}$ , and the lethargy gain per collision,  $\Sigma$ , values of both having been determined from differential scattering data<sup>13</sup> which were corrected to the center-of-mass system. Estimates were also made of the number of neutrons scattered to energies below the threshold of the U<sup>238</sup> fission detector used in the differential scattering experiment and the data were corrected accordingly. These corrected data could not be well approximated by the simple addition of a p-scattering term, although it was possible to evaluate  $\overline{\mu_0}$ , the average cosine of the scattering angle. Values of  $\sigma_{tr}$  and  $\Sigma$  were then calculated from  $\overline{\mu_0}$ .

Numerically integrating the corrected curve for differential scattering yielded the value

$$\vec{\mu}_{0} = \frac{\int_{\mu}^{\pi} \mu \sigma(\mu) d\mu}{\int_{\mu}^{\pi} \sigma(\mu) d\mu} = 0.254$$

for the average cosine of the scattering angle in the center-of-mass system with anisotropic scattering. Similarly, the value of  $\overline{\gamma}_0$ , the average cosine of the scattering angle in the laboratory system with anisotropic scattering, was found to be 0.329. This is to be compared with a value of  $\overline{\gamma}_0 = 0.074$  for the laboratory system with isotropic scattering.

The transport cross section  $\sigma_{\rm tr}$  for the system with anisotropic scattering found by

$$\sigma_{\rm tr} = (1 - \overline{\gamma}_0) \sigma_{\rm s}$$

is thus smaller than its isotropic value by the factor

$$\frac{1-0.329}{1-0.074} = 0.725$$

<sup>13.</sup> E. T. Jurney, "Inelastic Collision and Transport Cross Sections for Some Light Elements," IA-1339 (Dec., 1951).

The calculation of the lethargy gain per collision,  $\boldsymbol{\xi}$ , is presented in Appendix E and found to give a value of 0.150 for the system with anisotropic scattering. The corresponding value for the isotropic system is 0.206, the ratio of the two values being 0.728.

The product of the two ratios for  $\sigma_{tr}$  and  $\mathcal{F}$  then is

$$(0.725)(0.728) = 0.527$$

and this was assumed to be the factor which should be used in the calculations to correct the product  $\xi \sigma_{tr}$  to include anisotropic scattering in beryllium.

In order to test the validity of using this factor to include anisotropic scattering in the calculations, a calculation of the neutron age to thermal energy in beryllium was performed, the result of which could be compared with the accepted experimental value of  $\gamma$ . The relation used for this calculation of the age was

$$\mathcal{T} = \sum_{i=1}^{cn} \frac{\Delta u_i}{\Im \sum_{s}^{i} \xi_i \sum_{tr}^{i}}$$

where

Δ u<sub>i</sub> = lethargy width of energy group i,
 Σ<sup>i</sup><sub>s</sub> = macroscopic scattering cross section for energy group i,
 ℱ<sub>i</sub> = lethargy gain per collision in group i as defined in the preceding paragraph,

$$\Sigma_{+r}^{i}$$
 = macroscopic transport cross section for energy group i.

When only isotropic scattering in beryllium was assumed, the value of  $\gamma$  calculated with this relation, using the fission source distribution and the cross section values from Appendix C, was 84.3 cm<sup>2</sup>. The accepted experimental value is 98 cm<sup>2</sup>. Anisotropic scattering was then included in the calculation in an attempt to approach the accepted value. This was done by multiplying the term  $\Im_{\Sigma_{\rm tr}}$  in the upper energy range by the factor 0.527 determined in the preceding paragraph. The resulting calculated value of  $\gamma$  was 125 cm<sup>2</sup>, which indicated a gross over-correction for anisotropic scattering.

This discrepancy in the age calculation is an obvious indication of either poor fundamental data or an inadequate calculation method. Approximately two-thirds of the value of the age is contributed by terms in the summation above 1 kev of neutron energy, so uncertainties in fission source distribution or cross sections in this range have an exaggerated effect.

If a correction factor of 0.762 instead of 0.527 had been used, the calculated value of  $\gamma$  would have been in agreement with the accepted value.

This higher value of the correction was the one actually used in Methods III through VII to account for anisotropic scattering in beryllium.

Nuclear Reactions in Beryllium. A number of nuclear reactions with beryllium, such as (n,2n),  $(\alpha,n)$ ,  $(\gamma,n)$ , and inelastic scattering, are known to take place. Experimental measurements of cross sections for any of these events are very difficult, and the results are questionable. An attempt to measure the cross section for the (n,2n) reaction resulted in an illogical range of values from  $0.24 \pm 0.07$  barn, using a polonium-beryllium source, to  $-0.16 \pm 0.13$  barn, using a mock fission source;<sup>14</sup> therefore, no effort was made to consider this reaction in the present calculations. Neither are there any convenient methods for including the effects of the  $(\alpha,n)$  and  $(\gamma,n)$  reactions; however, these effects are probably small.

A value of 0.38 barn for the inelastic scattering cross section for neutrons above 1 Mev has been reported,  $^{13}$  which was found by subtracting the differential scattering cross sections integrated over a sphere from the total cross section. However, applying the corrections mentioned in the previous section to the observed data reduces the value of the cross section to 0.1 barn, which is so small that it may be completely obscured by experimental uncertainties. Nevertheless, it was included for one calculation of CA-1 (Method VI).

<u>Methods of</u> <u>Computing</u> <u>Neutron Slowing-Down Density</u>. With the exception of Method VII, the value of q, the neutron slowing-down density, was assumed to be equal to  $q = \emptyset \in \mathbb{Z}_{+}$ . In Method VII the Fermi slowing-down form,  $q = \emptyset \in \mathbb{Z}_{s}$ , was used. The scattering and total cross sections  $\mathcal{Z}_{s}$  and  $\mathcal{Z}_{t}$  included a leakage term DB<sup>2</sup>.

Correction for Effect of External Aluminum Grid. An estimate of the effect of the external grid was made in Method V by calculating the return probability of leakage neutrons. The calculational procedure is given in detail in Appendix F. The distribution of first collisions of leakage neutrons in the external grid is calculated for a particular energy group i. Isotropic scattering in the grid is assumed and the probability of a scattered neutron being returned to the core is calculated. The convolution of first collision density and the return probability gives the total probability of a leakage neutron being returned. The returning neutrons are assumed to be distributed uniformly throughout the core such that their importance is 0.535 times that of a normal mode source distribution. The effective number of leakage neutrons of groups i which are returned to the core are added to the next lower, or i + 1, source term. This allows some slowing down due to the collision with aluminum and affords a convenient means of including the effect in the bare reactor multigroup procedure. Counting only first collisions gives a result for the number of scattering collisions which is too low. On the other hand, this assumption is somewhat self-compensating in that it allows the neutrons to return unattenuated from the scattering points. The assumption of isotropic scattering and the uniform distribution of the returning neutrons overestimate the effectiveness of the aluminum grid so that the result is an upper value.

14. H. M. Agnew, "Measurement of  $\sigma(n, 2n)$  of Be<sup>9</sup>," LA-1371 (Mar., 1952).

#### Results of the Multigroup Calculations.

The results of the various calculations described above are summarized in Table 9. Variations in the results caused by individual corrections are shown in Table 10. A qualitative picture representing the effect of some of the corrections may be seen in Figs. 15, 16, and 17 which are plots of the leakage, absorption, and fission distributions in CA-1.

The predominant variation in leakage (Fig. 15) occurs in the upper energy range and results from the inclusion of anisotropic scattering in beryllium. With anisotropic scattering the high-energy mean free paths are greater, resulting in an increased leakage in this region. As a result of increased high-energy leakage, fewer neutrons are available for the lower energy range. This is obvious from the absorption and fission distributions of Figs. 16 and 17.

Self-shielding corrections have the effect of shifting the spectrum toward the lower energy. With no self-shielding corrections, a greater fraction of neutron absorption will be indicated in the intermediate range, or just above thermal (around u = 18), and fewer neutrons will be available to the thermal group. On the other hand, self-shielding corrections cause relatively small changes in the total number of absorptions or fissions.

As was noted in Section II, each of the assemblies was critical with a control rod partially removed so that the clean assemblies described had values of  $k_{eff}$  slightly greater than one. The total variation in  $k_{eff}$  for all the cases reported in Table 10 is 14%. This is actually quite arbitrary and might easily be made to vary much more by means of a particular choice of data.

For example, the correction made for anisotropic scattering was adjusted to fit the thermal neutron age, but it is smaller than experimental observations of differential scattering would indicate. Using the correction 0.527, which was indicated by the experimental observations, instead of the adjusted value of 0.762, would change the value of  $k_{eff}$  as much as 20% from the isotropic case. On the other hand, the inclusion of the (n,2n) reaction might completely overbalance this negative correction. For example, an assumed cross section of 0.01 barn for the (n,2n) reaction above 0.3 Mev would introduce a change in  $k_{eff}$  around 4% in either CA-1 or CA-18. This assumed 0.01 barn may be compared to an experimental uncertainty as great as 0.10 barn.

Some of the multigroup results were used to calculate cadmium fractions for indium and uranium and neutron self-shielding factors in the fuel. These particular calculations are relatively insensitive to high-energy effects and show neutron energy distribution shifts, such as those caused by different self-shielding corrections. The calculated results are compared to experimentally determined quantities in Table 11. Experimental values for cadmium fractions vary widely due to the heterogeneity of the system as shown in Fig. 10. While no exact comparison can be made due to this variation, the calculated values for CA-18 are generally low, even for the  $P_0$  self-shielding

			CA-1				CA-18		
		Leakage Fraction	Absorption Fraction	n <sup>k</sup> eff	Percent Thermal Fission	Leakage Fraction	Absorption Fraction	<sup>k</sup> eff	Percent Thermal Fission
Calculatio	ns								
Method I		0.4583	0.5417	1.0215	4.32	0.4560	0.5439	1.0550	37.95
Method I	I	0.4655	0.5345	1.0102 (	8.07	0.4843	0.5157	0.9839	45.32
Method I	Ia	0.4651	0.5349	1.0110	8.08	-	-	-	-
Method I	II	0.5094	0.4906	0.9290	7.95	0.5202	0.4798	0.9153	45.10
Method I	V	0.5074	0.4926	0.9323	6.14	0.5090	0.4910	0.9369	42.10
Method V	•	0.4911	0.5087	0.9632	6.19	0.4954	0.5046	0.9734	41.8
Method V	I	0.4872	0.5128	0.9675	6.17	-	-	-	-
Method V	II	0.5133	0.4865	0.9190	3.89	-	-	-	-
Experiment	,	-	-	1.0054	-	-	-	1.0020	-

Table 9. Summary of Results of Multigroup Calculations for CA-1 and CA-18: Comparison of Calculated k<sub>eff</sub> Values with Experimental k<sub>eff</sub> Values

Table 10.	Changes in Values of k for CA-1 and CA-18 Caused
	by Variations in Calculational Methods

		∠k
Variation	CA-1	CA-18
No correction for uranium self shielding to a P <sub>O</sub> correction (Method I to Method II)	-0.01133	-0.07118
P <sub>0</sub> correction for uranium self shielding to a P <sub>1</sub> correction (Method III to Method IV)	-0.00803	-0.04954
No inelastic scattering in U <sup>235</sup> to inelastic scattering for energies above 1 Mev (Method II to Method IIa)	+0.00081	
Isotropic scattering in beryllium		
energies above 1 Mev (Method IIa to Method III)	-0.08120	-0.06858
No inelastic scattering in beryllium to inelastic scattering for energies above 1 Mey (Method TV		
to Method VI)	+0.03519	
$\emptyset \mathbf{\xi} \Sigma_t$ slowing-down model to $\emptyset \mathbf{\xi} \Sigma_s$ Fermi model (Method IV to Method VII)	+0.0133	
No correction for effect of aluminum grid to a correction (Method IV to Method V)	+-0_03093	+0.03647



Fig. 15. Calculated Neutron Leakage Spectra for CA-1 for Various Assumptions and Corrections.



Fig. 16. Calculated Neutron Absorption Spectra for CA-1 for Various Assumptions and Corrections.



Fig. 17. Calculated Fission Spectra for CA-1 for Various Assumptions and Corrections.

	(	Cadmium Fractions			
	Fuel	L Disks	Indium Foils,	Self-Shielding Factor,	
	CA-1	CA-18	CA-18	CA-18	
Calculations			····		
Method I		0.753	0.258	0.649	
Method II	0.456	0.820	0.348	0.799	
Method III		0.816	0.326		
Method IV		0.793	0,289		
Method V		0.793	0.292	0.847	
Experiment	0.46	0.852 <b>-</b> 0.888	0.28 <b>-</b> 0.40	0.892	
		,	•		

Table 11. Comparison of Calculated Cadmium Fractions and Uranium Self-Shielding Factors with Experimental Values

correction. This would imply a more thermal system than the calculations show and perhaps a self-shielding effect even greater than that indicated by the  $P_0$  factors. One measurement of the cadmium fraction was made in CA-1, but the result is very approximate due to uncertainties in counting corrections available at that time.

An approximate method was used to calculate reactivity coefficients of poisons, but only in hopes of obtaining agreement to within an order of magnitude. Dependable self-shielding corrections are not available for the particular sizes of poison samples used. Further, a significant comparison could not be made without first calculating the complete adjoint flux corresponding to each of the multigroup calculations.

#### Comparison of Present Calculations for CA-1 with Previous Calculations

The method used in the original calculation<sup>15</sup> for CA-1 assumed the Fermi age form for the neutron slowing-down density but otherwise was similar to the second calculation reported here. The difference in  $k_{eff}$  primarily is due to differences in the high-energy scattering cross sections for beryllium which at that time were somewhat lower than values from more recent data.

<sup>15.</sup> A. O. Mooneyham, NEPA-1710, op. cit.

Following the experiment with CA-1, a number of other calculations<sup>16</sup> were made using a variety of assumptions and methods. Variations, such as using from 15 to 45 energy groups, assuming a monoenergetic 2-Mev fission neutron source instead of a fission neutron distribution, and varying the fuel self-shielding correction, gave only small changes in the calculated critical mass. However, it was found that a small increase in the scattering cross section for beryllium in the range 107 to 10<sup>5</sup> ev would give the correct value of critical mass. It was concluded that the calculations were very sensitive to high-energy scattering cross-section values, but were relatively insensitive to self-shielding corrections or the number of energy groups used.

In a later calculation<sup>17</sup> a correction for anisotropic scattering in beryllium was made in order to get a result for the age in agreement with the experimentally determined value. Several other corrections were made which were adequate to bring the calculation into agreement with the experimental size of CA-1. The agreement between this calculation and the experiment is considerably better than that found in the present set of calculations although it is somewhat artificial. As has been observed previously,  $k_{eff}$  is very sensitive to any change in high-energy data or any correction which effects the competition between high-energy leakage and slowing down. In the meantime, the same factors cause similar changes in calculated age. Mills did make a scattering correction in order to give the correct experimental age resulting in a decrease in  $k_{eff}$  of about 10%. The beryllium inelastic scattering correction approximately compensated  $k_{eff}$  for this loss, but its effect on age was not considered. In other words a consistent agreement with the experimental values of age and  $k_{eff}$  could not be found simultaneously.

This dilemma has not been resolved by the present investigation. Good agreement, however, may be found for  $k_{eff}$  by simply adjusting data in the high-energy range. While this is quite arbitrary, the choice may certainly be well within the range of uncertainty in fundamental data.

#### IV. CONCLUSIONS

The primary purpose of this report is to make available the experimental results obtained from the two beryllium-moderated critical assemblies. The multigroup calculations were made in an effort to evaluate some of the uncertainties involved and, if possible, to indicate an approach which would be applicable to other reactors. While it is possible by a judicious choice of data to get good agreement with the critical experiments, the rules governing the choice are certainly not clear.

An agreement in the value of  $k_{eff}$  does not in itself imply an understanding of the detailed neutron behavior. If, however, the critical size or critical mass of the system is the primary consideration, a fairly

17. C. B. Mills et al., ORNL-1493, op. cit.

<sup>16.</sup> G. M. Safanov, YF-10-45, op. cit.

dependable calculation may be made by adjusting high-energy fundamental data to give agreement with the critical experiments. A number of such quantities may be adjusted, namely, the value of the high-energy scattering cross section, the term for anisotropic scattering, inelastic scattering, etc. The critical size, or the actual  $k_{eff}$ , are convenient experimental quantities for a theoretical comparison since they may be observed accurately. Unfortunately, other available observed quantities are themselves subject to interpretations which involve uncertainties comparable to those in the multigroup calculations.

#### Appendix A

#### CALCULATION OF THE REACTIVITY VALUE OF A BERYLLIUM LAYER ON TOP OF CA-18

As stated in Section II, for the absolute calibration of the control rods in CA-18 it was necessary to determine the change in reactivity in the system effected by the addition of a 1/4-in.-thick layer of beryllium on the top of the assembly. The calculation was performed in the following manner. In a parallelepiped of augmented dimensions a x b x c, the geometric buckling of the system is

$$B_{o}^{2} = \left(\frac{\pi}{a}\right)^{2} + \left(\frac{\pi}{b}\right)^{2} + \left(\frac{\pi}{c}\right)^{2}$$
(A.1)

If c is increased a small amount, Sc, the buckling will be decreased an amount

$$\delta B^2 = -\left(\frac{\pi}{c}\right) \quad \left(\frac{2\delta c}{c}\right) \tag{A.2}$$

Assume that the multiplication keff may be written in the form

$$k_{eff} = AP_{n1} \tag{A.3}$$

1. ->

where A is a constant and  $P_{nl}$  is the nonleakage probability

$$P_{nl} = e^{-M^2 B_0^2}$$
(A.4)

 $M^2$  being the migration area. The change,  $\delta k$ , in  $k_{eff}$  found by substituting Eqs. A.2 and A.4 into A.3 is

 $k_{eff} + \delta k = Ae^{-M^2(B_0^2 + \delta B^2)}$ 

and

$$\frac{\delta k}{k_{\text{eff}}} = M^2 \left(\frac{\pi}{c}\right)^2 \left(\frac{2\delta c}{c}\right)$$
(A.5)

Solving Eq. A.4 for M<sup>2</sup> and substituting into Eq. A.5 gives:

$$\frac{\delta_k}{k_{eff}} = \left(\frac{1}{B_0^2}\right) \left(\frac{\pi}{c}\right)^2 \left(\frac{2\delta c}{c}\right) \ln \frac{1}{P_{nl}}$$
(A.6)

In Eq. A.6 the value of c is 73.89 cm which corresponds to one of the augmented dimensions in CA-18, and  $\sigma_c$  is 0.635 cm, which corresponds to the 1/4-in. thickness of beryllium. The values of  $B_0^2$  and  $P_{nl}$  were obtained from a multigroup calculation as follows:

Substituting these values in Eq. A.6 yields

$$\frac{k}{k_{eff}} = 0.003231.$$

#### Appendix B

#### MULTIGROUP FORMULATION\*

For the multigroup formulation the neutron energy range is divided into n groups plus a thermal group. Lethargy is defined as  $u \equiv ln(E_0/E)$ , where  $E_0 = 10^7$  ev and E is some particular neutron energy.

Consider group i in which u has the range  $u_{i-1} \le u \le u_i$ , and  $U_i = u_i - u_{i-1}$ . The fission source in group i, i.e., the number of neutrons born with energy i, is represented by  $\mathbb{Z}Z_i$ . Fission sources are normalized such that  $\sum_{i=1}^{n} \mathbb{Z}Z_i = \mathbb{Z}$ . The slowing-down density at a particular lethargy u is designated by q and the appropriate subscript. Cross sections are lethargy averages over each group and are considered as constant over the particular group. The probability that a neutron will slow down from u past  $u_i$  without loss due to absorption or leakage is given by

$$= \frac{2C_{i}(u_{i} - u)}{U_{i}}$$

$$= \frac{(\Sigma_{ai} + DB_{i}^{2})U_{i}}{(B.1)}$$

where

$$C_{i} = \frac{(\Sigma_{ai} + DB_{i}^{2})U_{i}}{2 \varepsilon_{i} \varepsilon_{ti}}$$
(B.1)

where

In the Fermi age approximation,  $\Sigma_t = \Sigma_s$ .

The contribution of  $q_i$  from group i-l is  $q_{i-l}e^{-2C_i}$ , while the contribution from the fission source is

$$\int_{u_{i-1}}^{u_i} Z_i \frac{du}{U_i} e^{-\frac{2C_i(u_i - u)}{U_i}} = \frac{Z_i}{2C_i} (1 - e^{-2C_i})$$
(B.2)

The total contribution is

$$q_i = q_{i-1} e^{-2C_i} + \frac{\frac{z}{Z_i}}{2C_i} (1 - e^{-2C_i})$$

<sup>\*</sup> This is a conventional multigroup formulation included here for completeness of the report; see, for example, reference 6, p. 115.

$$= \frac{1}{e^{C_{1}}} \left[ q_{1-1} e^{-C_{1}} + \frac{\frac{\sqrt{2}}{2C_{1}}}{2C_{1}} (e^{C_{1}} - e^{-C_{1}}) \right]$$

$$\cong \frac{q_{1-1} (1 - C_{1}) + \frac{\sqrt{2}}{2C_{1}}}{1 + C_{1}}$$

This is the recursion relation used in the multigroup calculations. If  $\phi_i$  is the average flux in group i, then the losses by leakage and absorption are

$$\mathbf{E}_{i} = \phi_{i} \mathbf{D} \mathbf{B}^{2} \mathbf{U}_{i} \text{ and } \mathbf{A}_{i} = \phi_{i} \boldsymbol{\Sigma}_{ai} \mathbf{U}_{i}$$

respectively. The neutron balance for group i is given by

$$q_{i-1} + \gamma Z_i = q_i + \phi_i U_i (\Sigma_{ai} + DB_i^2)$$
 (B.3)

Solving Eq. B.3 for  $\phi_i$  and using the definition of  $C_i$ ,

$$\phi_{i} = \frac{q_{i}}{\xi_{i}\Sigma_{ti}}$$
(B.4)

where

$$\overline{q}_{i} = \frac{q_{i-1} + q_{i}}{2}$$

The thermal-neutron flux is given by

$$\phi_{\text{th}} = \frac{q_{\text{n}}}{(\Sigma_{\text{a}}^{\text{th}} + DB_{\text{th}}^2)}$$
(B.5)

The fission rate in group i is  $F_i = \phi_i \Sigma_{fi} U_i$  where  $\Sigma_f$  is the fission cross section. The criticality equation is

$$\sum_{i=1}^{n} \mathbf{F}_{i} + \mathbf{F}_{th} = 1$$
 (B.6)

Inelastic scattering is included in the calculation by adding the neutrons scattered into a particular group to the fission source term in that group. The definition of  $C_1$  in Eq. B.1 is changed to include the inelastic scattering cross section by adding the inelastic cross section to the term in parentheses.

Upper Lettargy Humber digitLower (mit) $Z_{1,1}$ Number of $U_{10,2}^{(m)}$ Number of Number of $U_{10,2}^{(m)}$ Enriched Uranium $\sigma_8$ Beryllium $\sigma_8$ Alumium $\sigma_8$ Scattering $\sigma_8$ Cross Section <sup>C</sup> (cm <sup>-1</sup> )10.56.06 x 106 0.05250.05253.51.251.181.71.80.00010.266921.05.66 x 106 0.05250.26603.71.281.171.92.30.000250.252531.52.23 x 106 0.000350.5775k.21.221.151.93.00.000380.232552.50.82 x 1050.45775k.21.221.153.13.10.000380.235552.50.82 x 1050.45775k.21.221.153.13.10.000380.231663.0k.96 x 1050.459757.21.411.263.7k.00.000360.2355101.03 x 1020.069757.81.6531.40k.2k.00.00010.30097.09.12 x 10201.52.2315.606.01.350.00160.7701111.41.12 x 10 <sup>2</sup> 01.52.231.566.01.350.00150.8801313.41.51 x 10 <sup>2</sup> 01.252.2315.606.01.350.00160.7701212.63.772 x 10 <sup>1</sup> 09.55.14.0.3 <t< th=""><th colspan="3">Group</th><th colspan="5">Microscopic Cross Sections (barns)</th><th colspan="2">Stainless Steel Macroscopic</th></t<>	Group			Microscopic Cross Sections (barns)					Stainless Steel Macroscopic		
$\begin{array}{c c c c c c c c c c c c c c c c c c c $		Upper Letheray	Lower	ZZ <sub>1</sub> , Number of	Enr	iched Ure	กว่าเพื่	Berry 11 jum		1 01100	Scattering
$ \begin{array}{c} 1 & 0.5 & 6.06 \times 10^6 & 0.0525 & 3.5 & 1.25 & 1.18 & 1.7 & 1.8 & 0.0001 & 0.269 \\ 2 & 1.0 & 3.68 \times 10^6 & 0.2600 & 3.7 & 1.24 & 1.17 & 1.9 & 2.3 & 0.0002 & 0.303 \\ 3 & 1.5 & 2.23 \times 10^6 & 0.5175 & 4.3 & 1.23 & 1.16 & 2.4 & 2.9 & 0.00035 & 0.252 \\ 4 & 2.0 & 1.35 \times 10^5 & 0.5775 & 4.2 & 1.22 & 1.15 & 1.9 & 3.0 & 0.00038 & 0.235 \\ 5 & 2.5 & 0.82 \times 10^5 & 0.4575 & 4.8 & 1.21 & 1.13 & 3.1 & 3.1 & 0.00038 & 0.201 \\ 6 & 3.0 & 4.96 \times 10^5 & 0.2925 & 6.0 & 1.26 & 1.18 & 4.0 & 3.5 & 0.0006 & 0.201 \\ 7 & 3.5 & 3.02 \times 10^5 & 0.1675 & 7.2 & 1.41 & 1.26 & 3.7 & 4.0 & 0.0004 & 0.294 \\ 8 & 4.0 & 1.85 \times 10^5 & 0.0675 & 9.44 & 2.78 & 2.26 & 5.1 & 4.0 & 0.0003 & 0.350 \\ 9 & 7.0 & 9.12 \times 10^2 & 0 & 10.5 & 0.606 & 6.23 & 6.0 & 1.35 & 0.0006 & 0.570 \\ 10 & 10.0 & 4.54 \times 10^2 & 0 & 15 & 22.3 & 15.60 & 6.0 & 1.35 & 0.0006 & 0.770 \\ 11 & 11.4 & 1.12 \times 10^2 & 0 & 15 & 22.3 & 15.60 & 6.0 & 1.355 & 0.0046 & 0.770 \\ 12 & 12.6 & 3.372 \times 10^1 & 0 & 12 & 47.0 & 33.4 & 6.0 & 1.355 & 0.0045 & 0.680 \\ 13 & 13.4 & 1.515 \times 10^1 & 0 & 9.0 & 61.4 & 43.8 & 6.0 & 1.35 & 0.0046 & 0.680 \\ 14 & 13.8 & 1.005 \times 10^{-1} & 0 & 8.5 & 26.5 & 18.7 & 6.0 & 1.355 & 0.0048 \\ 15 & 14.6 & 4.546 & 0 & 8.5 & 22.9 & 32.7 & 6.0 & 1.355 & 0.0048 \\ 16 & 15.8 & 1.375 & 10^{-1} & 0 & 8.5 & 76.5 & 60.1 & 6.35 & " & 0.880 \\ 16 & 15.8 & 1.375 & 10^{-1} & 0 & 8.5 & 26.5 & 18.7 & 6.0 & 1.355 & " & 0.880 \\ 17 & 16.2 & 9.214 \times 10^{-1} & 0 & 8.5 & 76.5 & 60.1 & 6.0 & 1.355 & " & 0.880 \\ 18 & 16.6 & 6.176 \times 10^{-1} & 0 & 8.5 & 273 & 216.7 & 5.5 & 1.355 & " & 0.880 \\ 19 & 17.0 & 4.140 \times 10^{-1} & 0 & 8.5 & 273 & 216.7 & 5.5 & 1.355 & " & 0.880 \\ 20 & 17.4 & 2.775 \times 10^{-1} & 0 & 8.5 & 273 & 216.7 & 5.5 & 1.355 & " & 0.880 \\ 22 & 17.4 & 2.275 \times 10^{-1} & 0 & 8.5 & 273 & 216.7 & 5.5 & 1.355 & " & 0.880 \\ 23 & 18.0 & 1.523 \times 10^{-2} & 0 & 8.5 & 273 & 216.7 & 5.2 & 1.355 & " & 0.880 \\ 24 & 18.4 & 8.36 \times 10^{-2} & 0 & 8.5 & 273 & 216.7 & 5.2 & 1.355 & " & 0.880 \\ 25 & 18.4 & 1.021 \times 10^{-2} & 0 & 8.5 & 273 & 216.7 & 5.2 & 1.355 & " & 0.880 \\ 25 & 18.4 & 1.021 \times 10^{-2} & 0 & 8.5 &$	Number	Limit	Limit (ev)	Neutronsa	σ <sub>s</sub>	σ <sub>a</sub>	σ <sub>f</sub>	σ <sub>s</sub>	σ <sub>s</sub>	J. J	(cm <sup>-1</sup> )
$ \begin{array}{cccccccccccccccccccccccccccccccccccc$		0=0	(E0=101 6V)							u	
$ \begin{array}{cccccccccccccccccccccccccccccccccccc$	1	0.5	$6.06 \times 10^{6}$	0.0525	3.5	1.25	1.18	1.7	1.8	0.0001	0.269
$ \begin{array}{cccccccccccccccccccccccccccccccccccc$	2	1.0	$3.68 \times 10^{6}$	0.2600	3.7	1.24	1.17	1.9	2.3	0.0002	0.303
$ \begin{array}{cccccccccccccccccccccccccccccccccccc$	3	1.5	$2.23 \times 10^{6}$	0.5175	4.3	1.23	1.16	2.4	2.9	0.00035	0.252
$ \begin{array}{cccccccccccccccccccccccccccccccccccc$	<u>4</u>	2.0	$1.35 \times 10^{6}$	0.5775	4.2	1.22	1.15	1.9	3.0	0.00038	0.235
$ \begin{array}{cccccccccccccccccccccccccccccccccccc$	5	2.5	$0.82 \times 10^{2}$	0.4575	4.8	1.21	1.13	3.1	3.1	0.00038	0.201
73.53.02 x $10^{2}$ 0.16757.21.411.263.74.00.00040.29484.01.83 x $10^{5}$ 0.08757.81.631.404.24.00.00140.30097.09.12 x $10^{2}$ 0.08759.42.782.265.14.00.00300.5701010.04.54 x $10^{2}$ 010.58.886.236.01.350.00090.5701111.41.12 x $10^{2}$ 01522.315.606.01.350.00140.7701212.63.77 x $10^{1}$ 01247.035.46.01.350.00450.8801313.41.515 x $10^{1}$ 09.558.140.36.01.35"0.8801415.81.015 x $10^{1}$ 09.558.140.36.01.35"0.8801514.64.54608.552.932.76.01.35"0.8801615.8 $1.775$ 08.576.56.01.35"0.8801616.66.176 x $10^{-1}$ 08.578.374.45.81.355"0.8801917.04.140 x $10^{-1}$ 08.5227.174.45.81.355"0.8802117.62.275 x $10^{-1}$ 08.5237.18.71.35"0.8802217.4 <td< td=""><td>6</td><td>3.0</td><td><math>4.98 \times 10^{5}</math></td><td>0.2925</td><td>6.0</td><td>1.26</td><td>1.18</td><td>4.0</td><td>3.5</td><td>0.0006</td><td>0.201</td></td<>	6	3.0	$4.98 \times 10^{5}$	0.2925	6.0	1.26	1.18	4.0	3.5	0.0006	0.201
	7	3.5	3.02 x 10 <sup>5</sup>	0.1675	7.2	1.41	1.26	3.7	4.0	0.0004	0.294
97.0 $9,12 \times 10^3$ $0.0875$ $9.4$ $2.78$ $2.26$ $5.1$ $4.0$ $0.0030$ $0.350$ 10 $10.0$ $4.54 \times 10^2$ $0$ $10.5$ $0.88$ $6.23$ $6.0$ $1.35$ $0.0009$ $0.570$ 11 $11.4$ $1.12 \times 10^2$ $0$ $15$ $22.3$ $15.60$ $6.0$ $1.35$ $0.0016$ $0.770$ 12 $12.6$ $3.372 \times 10^1$ $0$ $12$ $47.0$ $33.4$ $6.0$ $1.35$ $0.0043$ $0.880$ 13 $13.4$ $1.515 \times 10^1$ $0$ $9.5$ $58.1$ $40.3$ $6.0$ $1.35$ $n.0043$ $0.880$ 14 $13.6$ $1.015 \times 10^1$ $0$ $9.0$ $61.4$ $43.8$ $6.0$ $1.35$ $n.880$ 15 $14.6$ $h.546$ $0$ $8.5$ $22.9$ $32.7$ $6.0$ $1.35$ $n.880$ 16 $15.8$ $1.775$ $0$ $8.5$ $26.3$ $18.7$ $6.0$ $1.35$ $n.880$ 16 $16.6$ $6.176 \times 10^{-1}$ $0$ $8.5$ $78.5$ $60.1$ $1.35$ $n.880$ 18 $16.6$ $6.176 \times 10^{-1}$ $0$ $8.5$ $78.3$ $74.4$ $5.8$ $1.35$ $n.880$ 20 $17.4$ $2.775 \times 10^{-1}$ $0$ $8.5$ $227.1$ $174.6$ $5.5$ $1.35$ $n.880$ 21 $17.6$ $2.275 \times 10^{-1}$ $0$ $8.5$ $273$ $184.7$ $5.5$ $1.35$ $n.880$ 22 $17.8$ $1.860 \times 10^{-2}$	8	4.0	1.83 x 10 <sup>5</sup>	0.0875	7.8	1.63	1.40	4.2	4.0	0.0014	0.300
$ \begin{array}{cccccccccccccccccccccccccccccccccccc$	9	7.0	$9.12 \times 10^3$	0.0875	9.4	2.78	2.26	5.1	4.0	0.0030	0.350
1111.4 $1.12 \times 10^{6}$ 01522.315.606.0 $1.35$ 0.00160.7701212.6 $3.772 \times 10^{1}$ 012 $47.0$ $33.4$ 6.0 $1.35$ 0.00430.88013 $13.4$ $1.515 \times 10^{1}$ 09.5 $58.1$ $40.3$ 6.0 $1.35$ $Assume 1/v$ 0.88014 $13.8$ $1.015 \times 10^{1}$ 09.0 $61.4$ $43.8$ 6.0 $1.35$ "0.88015 $14.6$ $4.546$ 0 $8.5$ $52.9$ $32.7$ 6.0 $1.35$ "0.88016 $15.8$ $1.775$ 0 $8.5$ $26.3$ $18.7$ 6.0 $1.35$ "0.88016 $15.8$ $1.775$ 0 $8.5$ $76.5$ $60.1$ $6.0$ $1.35$ "0.88017 $16.2$ $9.214 \times 10^{-1}$ 0 $8.5$ $78.3$ $74.4$ $5.8$ $1.35$ "0.88019 $17.0$ $4.140 \times 10^{-1}$ 0 $8.5$ $172.$ $143.5$ $5.7$ $1.35$ "0.88020 $17.4$ $2.775 \times 10^{-1}$ 0 $8.5$ $237.$ $174.6$ $5.5$ $1.35$ "0.88021 $17.6$ $2.275 \times 10^{-1}$ 0 $8.5$ $273.$ $198.9$ $5.5$ $1.35$ "0.88022 $17.8$ $1.660 \times 10^{-2}$ 0 $8.5$ $273.$ $198.9$ $5.5$ $1.35$ "0.88023 $18.0$ $1.523 \times 10^{-2}$	10	10.0	$4.54 \times 10^2$	0	10.5	8.88	6.23	6.0	1.35	0.0009	0.570
12       12.66 $3.372 \times 10_1^1$ 0       12 $47.6$ $33.4$ 6.0 $1.55$ $0.0043$ $0.880$ 13       13.4 $1.515 \times 10^1$ 0       9.5 $58.1$ $40.3$ $6.0$ $1.35$ Assume $1/v$ $0.880$ 14       13.8 $1.015 \times 10^1$ 0       9.0 $61.4$ $43.8$ $6.0$ $1.35$ " $0.880$ 15 $14.6$ $4.546$ 0 $8.5$ $52.9$ $32.7$ $6.0$ $1.35$ <"	11	11.4	$1.12 \times 10^2$	0	15	22.3	15.60	6.0	1.35	0.0016	0.770
1313.4 $1.515 \times 10^{1}$ 09.5 $58.1$ $40.3$ 6.0 $1.35$ Assume $1/v$ 0.8801413.81.015 $\times 10^{1}$ 09.0 $61.4$ $45.8$ $6.0$ $1.35$ "0.8801514.6 $4.546$ 0 $8.5$ $52.9$ $32.7$ $6.0$ $1.35$ "0.8801615.8 $1.775$ 0 $8.5$ $26.3$ $18.7$ $6.0$ $1.35$ "0.8801615.8 $1.775$ 0 $8.5$ $76.5$ $60.1$ $6.0$ $1.35$ "0.8801816.6 $6.176 \times 10^{-1}$ 0 $8.5$ $76.5$ $60.1$ $6.0$ $1.35$ "0.8801917.0 $4.140 \times 10^{-1}$ 0 $8.5$ $78.3$ $74.4$ $5.8$ $1.355$ "0.8802017.4 $2.775 \times 10^{-1}$ 0 $8.5$ $227.179.5$ $5.6$ $1.355$ "0.8802117.6 $2.275 \times 10^{-1}$ 0 $8.5$ $237.179.5$ $1.355$ "0.8802217.8 $1.860 \times 10^{-1}$ 0 $8.5$ $237.179.5$ $1.355$ "0.8802318.0 $1.523 \times 10^{-1}$ 0 $8.5$ $257.3$ $1.355$ "0.88024 $18.2$ $1.247 \times 10^{-1}$ 0 $8.5$ $257.3$ $1.355$ "0.88025 $18.4$ $1.021 \times 10^{-1}$ 0 $8.5$ $273$ $216.7$ $5.2$ $1.355$ "0.8802	12	12.6	$3.372 \times 10^{1}$	0	12	47.0	33.4	6.0	1.35	0.0043	0.880
1413.8 $1.015 \times 10^{1}$ 09.0 $61.4$ $43.8$ $6.0$ $1.55$ " $0.880$ 1514.6 $4.546$ 0 $8.5$ $52.9$ $32.7$ $6.0$ $1.35$ " $0.880$ 1615.8 $1.375$ 0 $8.5$ $52.9$ $32.7$ $6.0$ $1.35$ " $0.880$ 17 $16.2$ $9.214 \times 10^{-1}$ 0 $8.5$ $76.5$ $60.1$ $6.0$ $1.35$ " $0.880$ 18 $16.6$ $6.176 \times 10^{-1}$ 0 $8.5$ $76.5$ $60.1$ $6.0$ $1.35$ " $0.880$ 19 $17.0$ $4.140 \times 10^{-1}$ 0 $8.5$ $78.3$ $74.4$ $5.8$ $1.35$ " $0.880$ 20 $17.4$ $2.775 \times 10^{-1}$ 0 $8.5$ $242.179.3$ $5.6$ $1.35$ " $0.880$ 21 $17.6$ $2.275 \times 10^{-1}$ 0 $8.5$ $227.174.6$ $5.5$ $1.35$ " $0.880$ 22 $17.8$ $1.860 \times 10^{-1}$ 0 $8.5$ $237.184.7$ $5.4$ $1.355$ " $0.880$ 23 $18.0$ $1.523 \times 10^{-1}$ 0 $8.5$ $237.184.7$ $5.4$ $1.355$ " $0.880$ 24 $16.2$ $1.247 \times 10^{-1}$ 0 $8.5$ $273$ $16.7$ $5.2$ $1.355$ " $0.880$ 24 $18.2$ $1.247 \times 10^{-2}$ 0 $8.5$ $306.9$ $5.5$ $1.355$ " $0.880$ 25 $18.4$ $1.021 \times 10^{-2}$ 0 $8.5$ </td <td>13</td> <td>13.4</td> <td><math>1.515 \times 10^{1}</math></td> <td>0</td> <td>9.5</td> <td>58.1</td> <td>40.3</td> <td>6.0</td> <td>1.35</td> <td>Assume</td> <td>1/v 0.880</td>	13	13.4	$1.515 \times 10^{1}$	0	9.5	58.1	40.3	6.0	1.35	Assume	1/v 0.880
1514.6 $4.546$ 0 $8.5$ $52.9$ $32.7$ $6.0$ $1.35$ " $0.880$ 16 $15.8$ $1.375$ 0 $8.5$ $26.3$ $18.7$ $6.0$ $1.35$ " $0.880$ 17 $16.2$ $9.214 \times 10^{-1}$ 0 $8.5$ $76.5$ $60.1$ $6.0$ $1.35$ " $0.880$ 18 $16.6$ $6.176 \times 10^{-1}$ 0 $8.5$ $76.5$ $60.1$ $6.0$ $1.35$ " $0.880$ 19 $17.0 \times 116 \times 10^{-1}$ 0 $8.5$ $78.3$ $74.4$ $5.8$ $1.35$ " $0.880$ 20 $17.4 \times 2.775 \times 10^{-1}$ 0 $8.5 \times 172.$ $143.5 \times 5.7$ $1.35$ " $0.880$ 21 $17.6 \times 2.275 \times 10^{-1}$ 0 $8.5 \times 242.$ $179.3 \times 5.6$ $1.355$ " $0.880$ 21 $17.6 \times 2.275 \times 10^{-1}$ 0 $8.5 \times 227.$ $174.6 \times 5.5$ $1.355$ " $0.880$ 22 $17.8 \times 1.860 \times 10^{-1}$ 0 $8.5 \times 273.$ $198.9 \times 5.5$ $1.355$ " $0.880$ 23 $18.0 \times 1.523 \times 10^{-1}$ 0 $8.5 \times 273.$ $198.9 \times 5.5$ $1.355$ " $0.880$ 24 $18.2 \times 1.247 \times 10^{-1}$ 0 $8.5 \times 305.$ $241.9 \times 5.0$ $1.355$ " $0.880$ 25 $18.4 \times 10^{-2}$ 0 $8.5 \times 305.$ $241.9 \times 5.0$ $1.355$ " $0.880$ 26 $18.6 \times 8.36 \times 10^{-2}$ 0 $8.5 \times 305.$ $241.9 \times 5.0$ $1.355$ " $0.880$ 29 $19.4 \times 5.765 \times 10^{-2}$ </td <td>14</td> <td>13.8</td> <td><math>1.015 \times 10^{1}</math></td> <td>0</td> <td>9.0</td> <td>61.4</td> <td>43.8</td> <td>6.0</td> <td>1.35</td> <td>n</td> <td>0.880</td>	14	13.8	$1.015 \times 10^{1}$	0	9.0	61.4	43.8	6.0	1.35	n	0.880
$16$ $15.8$ $1.575$ $0$ $8.5$ $26.5$ $18.7$ $6.0$ $1.55$ " $0.080$ $17$ $16.2$ $9.214 \times 10^{-1}$ $0$ $8.5$ $76.5$ $60.1$ $6.0$ $1.35$ " $0.880$ $18$ $16.6$ $6.176 \times 10^{-1}$ $0$ $8.5$ $76.5$ $60.1$ $6.0$ $1.35$ " $0.880$ $19$ $17.0$ $4.140 \times 10^{-1}$ $0$ $8.5$ $78.3$ $74.4$ $5.8$ $1.35$ " $0.880$ $20$ $17.4$ $2.775 \times 10^{-1}$ $0$ $8.5$ $172.$ $143.5$ $5.7$ $1.35$ " $0.880$ $21$ $17.6$ $2.275 \times 10^{-1}$ $0$ $8.5$ $242.$ $179.3$ $5.6$ $1.355$ " $0.880$ $21$ $17.4$ $2.275 \times 10^{-1}$ $0$ $8.5$ $227.$ $174.6$ $5.5$ $1.355$ " $0.880$ $22$ $17.8$ $1.860 \times 10^{-1}$ $0$ $8.5$ $237.$ $184.7$ $5.4$ $1.355$ " $0.880$ $23$ $18.0$ $1.523 \times 10^{-1}$ $0$ $8.5$ $273$ $198.9$ $5.3$ $1.355$ " $0.880$ $24$ $18.2$ $1.247 \times 10^{-1}$ $0$ $8.5$ $273$ $216.7$ $5.2$ $1.355$ " $0.880$ $26$ $18.6$ $8.36 \times 10^{-2}$ $0$ $8.5$ $302$ $241.9$ $5.0$ $1.355$ " $0.880$ $27$ $18.8$ $6.843 \times 10^{-2}$ $0$ $8.5$ $342$ $273.6$ $4.8$ <	15	14.6	4.546	0	8.5	52.9	32.7	6.0	1.35	11	0.880
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	16	15.8	1.375	0	8.5	26.3	18.7	6.0	1.35	11	0.880
$16$ $16.6$ $6.176 \times 10^{-1}$ $0$ $8.5$ $60.6$ $52.5$ $6.0$ $1.35$ " $0.880$ $19$ $17.0$ $4.140 \times 10^{-1}$ $0$ $8.5$ $78.3$ $74.4$ $5.8$ $1.35$ " $0.880$ $20$ $17.4$ $2.775 \times 10^{-1}$ $0$ $8.5$ $172.$ $143.5$ $5.7$ $1.355$ " $0.880$ $21$ $17.6$ $2.275 \times 10^{-1}$ $0$ $8.5$ $242.$ $179.3$ $5.6$ $1.355$ " $0.880$ $22$ $17.8$ $1.860 \times 10^{-1}$ $0$ $8.5$ $227.$ $174.6$ $5.5$ $1.355$ " $0.880$ $22$ $17.8$ $1.860 \times 10^{-1}$ $0$ $8.5$ $227.$ $174.6$ $5.5$ $1.355$ " $0.880$ $24$ $18.2$ $1.247 \times 10^{-1}$ $0$ $8.5$ $253$ $198.9$ $5.3$ $1.355$ " $0.880$ $25$ $18.4$ $1.021 \times 10^{-1}$ $0$ $8.5$ $273$ $216.7$ $5.2$ $1.355$ " $0.880$ $26$ $18.6$ $8.36 \times 10^{-2}$ $0$ $8.5$ $342$ $273.6$ $4.8$ $1.355$ " $0.880$ $27$ $18.8$ $6.843 \times 10^{-2}$ $0$ $8.5$ $386$ $311$ $4.5$ $1.355$ " $0.880$ $29$ $19.2$ $4.587 \times 10^{-2}$ $0$ $8.5$ $426$ $349.3$ $4.5$ $1.355$ " $0.880$ $30$ $19.4$ $3.756 \times 10^{-2}$ $0$ $8.5$ $530$ $495.3$ <td>17</td> <td>16.2</td> <td>9.214 x <math>10^{-1}</math></td> <td>0</td> <td>8.5</td> <td>76.5</td> <td>60.i</td> <td>6.0</td> <td>1.35</td> <td>Ħ</td> <td>0.880</td>	17	16.2	9.214 x $10^{-1}$	0	8.5	76.5	60.i	6.0	1.35	Ħ	0.880
1917.04.140 x 10 <sup>-1</sup> 08.5 $78.3$ $74.4$ 5.81.35"0.8802017.42.775 x 10 <sup>-1</sup> 08.5172.143.55.71.35"0.8802117.62.275 x 10 <sup>-1</sup> 08.5242.179.35.61.35"0.8802217.81.860 x 10 <sup>-1</sup> 08.5227.174.65.51.35"0.8802318.01.523 x 10 <sup>-1</sup> 08.5237.184.75.41.35"0.8802418.21.247 x 10 <sup>-1</sup> 08.5273198.95.31.35"0.8802418.21.247 x 10 <sup>-1</sup> 08.5273216.75.21.35"0.8802518.41.021 x 10 <sup>-1</sup> 08.5305241.95.01.35"0.8802618.68.36 x 10 <sup>-2</sup> 08.5305241.95.01.35"0.8802718.86.843 x 10 <sup>-2</sup> 08.53663114.51.35"0.8802919.24.587 x 10 <sup>-2</sup> 08.5426349.34.31.35"0.8803019.43.756 x 10 <sup>-2</sup> 08.5530441.83.51.35"0.8803119.63.075 x 10 <sup>-2</sup> 08.5590495.03.41.35"0.8803219.82.518 x 10 <sup></sup>	18	16.6	6.176 x 10 <sup>-1</sup>	0	8.5	60.6	<b>52.</b> 5	6.0	1.35	Ħ	0.880
$20$ $17.4$ $2.775 \times 10^{-1}$ $0$ $8.5$ $172.$ $143.5$ $5.7$ $1.355$ " $0.880$ $21$ $17.6$ $2.275 \times 10^{-1}$ $0$ $8.5$ $242.$ $179.3$ $5.6$ $1.355$ " $0.880$ $22$ $17.8$ $1.860 \times 10^{-1}$ $0$ $8.5$ $242.$ $179.3$ $5.6$ $1.355$ " $0.880$ $23$ $18.0$ $1.523 \times 10^{-1}$ $0$ $8.5$ $227.$ $174.6$ $5.5$ $1.355$ " $0.880$ $24$ $18.2$ $1.247 \times 10^{-1}$ $0$ $8.5$ $237.$ $184.7$ $5.4$ $1.355$ " $0.880$ $25$ $18.4$ $1.021 \times 10^{-1}$ $0$ $8.5$ $273$ $216.7$ $5.2$ $1.355$ " $0.880$ $26$ $18.6$ $8.36 \times 10^{-2}$ $0$ $8.5$ $305$ $241.9$ $5.0$ $1.355$ " $0.880$ $27$ $18.8$ $6.843 \times 10^{-2}$ $0$ $8.5$ $342$ $273.6$ $4.8$ $1.355$ " $0.880$ $28$ $19.0$ $5.603 \times 10^{-2}$ $0$ $8.5$ $326$ $311$ $4.5$ $1.355$ " $0.880$ $29$ $19.2$ $4.587 \times 10^{-2}$ $0$ $8.5$ $426$ $349.3$ $4.3$ $1.355$ " $0.880$ $30$ $19.4$ $3.756 \times 10^{-2}$ $0$ $8.5$ $570$ $441.8$ $3.5$ $1.355$ " $0.880$ $31$ $19.6$ $3.075 \times 10^{-2}$ $0$ $8.5$ $590$ $495.0$ <	19	17.0	4.140 x 10-1	0	8.5	78.3	74.4	5.8	1.35	ti	0.880
21 $17.6$ $2.275 \times 10^{-1}$ 0 $8.5$ $242.$ $179.3$ $5.6$ $1.35$ "0.88022 $17.8$ $1.860 \times 10^{-1}$ 0 $8.5$ $227.$ $174.6$ $5.5$ $1.35$ "0.88023 $18.0$ $1.523 \times 10^{-1}$ 0 $8.5$ $237.$ $184.7$ $5.4$ $1.35$ "0.88024 $18.2$ $1.247 \times 10^{-1}$ 0 $8.5$ $253$ $198.9$ $5.3$ $1.35$ "0.88025 $18.4$ $1.021 \times 10^{-1}$ 0 $8.5$ $273$ $216.7$ $5.2$ $1.35$ "0.88026 $18.6$ $8.36 \times 10^{-2}$ 0 $8.5$ $305$ $241.9$ $5.0$ $1.35$ "0.88026 $18.6$ $6.843 \times 10^{-2}$ 0 $8.5$ $342$ $273.6$ $4.8$ $1.35$ "0.88027 $18.8$ $6.843 \times 10^{-2}$ 0 $8.5$ $342$ $273.6$ $4.8$ $1.35$ "0.88028 $19.0$ $5.603 \times 10^{-2}$ 0 $8.5$ $346$ $311$ $4.5$ $1.35$ "0.88029 $19.2$ $4.587 \times 10^{-2}$ 0 $8.5$ $475$ $393.2$ $3.9$ $1.35$ "0.88030 $19.4$ $3.756 \times 10^{-2}$ 0 $8.5$ $530$ $441.8$ $3.5$ $1.35$ "0.88031 $19.6$ $3.075 \times 10^{-2}$ 0 $8.5$ $590$ $495.0$ $3.4$ $1.35$ "0.88032 $1$	20	17.4	2.775 x 10 <sup>-1</sup>	0	8.5	172.	143.5	5.7	1.35	**	0.880
22 $17.8$ $1.860 \times 10^{-1}$ 0 $8.5$ $227.$ $174.6$ $5.5$ $1.35$ " $0.880$ 23 $18.0$ $1.523 \times 10^{-1}$ 0 $8.5$ $237.$ $184.7$ $5.4$ $1.35$ " $0.880$ 24 $18.2$ $1.247 \times 10^{-1}$ 0 $8.5$ $253$ $198.9$ $5.3$ $1.35$ " $0.880$ 25 $18.4$ $1.021 \times 10^{-1}$ 0 $8.5$ $273$ $216.7$ $5.2$ $1.355$ " $0.880$ 26 $18.6$ $8.36 \times 10^{-2}$ 0 $8.5$ $305$ $241.9$ $5.0$ $1.355$ " $0.880$ 26 $18.6$ $6.843 \times 10^{-2}$ 0 $8.5$ $305$ $241.9$ $5.0$ $1.355$ " $0.880$ 27 $18.8$ $6.843 \times 10^{-2}$ 0 $8.5$ $342$ $273.6$ $4.8$ $1.355$ " $0.880$ 28 $19.0$ $5.603 \times 10^{-2}$ 0 $8.5$ $426$ $349.3$ $4.5$ $1.355$ " $0.880$ 29 $19.2$ $4.587 \times 10^{-2}$ 0 $8.5$ $426$ $349.3$ $4.5$ $1.355$ " $0.880$ 30 $19.4$ $3.776 \times 10^{-2}$ 0 $8.5$ $530$ $441.8$ $3.55$ $1.355$ " $0.880$ 31 $19.6$ $3.075 \times 10^{-2}$ 0 $8.5$ $590$ $495.0$ $3.4$ $1.355$ " $0.880$ $32$ $19.8$ $2.518 \times 10^{-2}$ 0 $8.5$ $590$ $495.0$ $3.4$ $1.355$ "	21	17.6	2.275 x 10 <sup>-1</sup>	0	8.5	242.	179.3	5.6	1.35	91	0.880
23 $18.0$ $1.523 \times 10^{-1}$ 0 $8.5$ $237.$ $184.7$ $5.4$ $1.35$ "0.88024 $18.2$ $1.247 \times 10^{-1}$ 0 $8.5$ $253$ $198.9$ $5.3$ $1.35$ "0.88025 $18.4$ $1.021 \times 10^{-1}$ 0 $8.5$ $273$ $216.7$ $5.2$ $1.35$ "0.88026 $18.6$ $8.36 \times 10^{-2}$ 0 $8.5$ $305$ $241.9$ $5.0$ $1.35$ "0.88027 $18.8$ $6.843 \times 10^{-2}$ 0 $8.5$ $342$ $273.6$ $4.8$ $1.35$ "0.88028 $19.0$ $5.603 \times 10^{-2}$ 0 $8.5$ $346$ $311$ $4.5$ $1.35$ "0.88029 $19.2$ $4.587 \times 10^{-2}$ 0 $8.5$ $426$ $349.3$ $4.3$ $1.35$ "0.88030 $19.4$ $3.756 \times 10^{-2}$ 0 $8.5$ $530$ $441.8$ $3.5$ $1.35$ "0.88031 $19.6$ $3.075 \times 10^{-2}$ 0 $8.5$ $590$ $495.0$ $3.4$ $1.35$ "0.880 $32$ $19.8$ $2.518 \times 10^{-2}$ 0 $8.5$ $590$ $495.0$ $3.4$ $1.35$ "0.88032 $19.8$ $2.518 \times 10^{-2}$ 0 $8.5$ $590$ $495.0$ $3.4$ $1.35$ "0.88034 $10.9$ $2.518 \times 10^{-2}$ 0 $8.5$ $590$ $495.0$ $3.4$ $1.35$ "0.880	22	17.8	$1.860 \times 10^{-1}$	0	8.5	227.	174.6	5.5	1.35	**	0.880
$24$ $18.2$ $1.247 \times 10^{-1}$ $0$ $8.5$ $253$ $198.9$ $5.3$ $1.35$ " $0.880$ $25$ $18.4$ $1.021 \times 10^{-1}$ $0$ $8.5$ $273$ $216.7$ $5.2$ $1.35$ " $0.880$ $26$ $18.6$ $8.36 \times 10^{-2}$ $0$ $8.5$ $305$ $241.9$ $5.0$ $1.35$ " $0.880$ $27$ $18.8$ $6.843 \times 10^{-2}$ $0$ $8.5$ $342$ $273.6$ $4.8$ $1.35$ " $0.880$ $28$ $19.0$ $5.603 \times 10^{-2}$ $0$ $8.5$ $346$ $311$ $4.5$ $1.35$ " $0.880$ $29$ $19.2$ $4.587 \times 10^{-2}$ $0$ $8.5$ $426$ $349.3$ $4.3$ $1.35$ " $0.880$ $30$ $19.4$ $3.756 \times 10^{+2}$ $0$ $8.5$ $475$ $393.2$ $3.9$ $1.35$ " $0.880$ $31$ $19.6$ $3.075 \times 10^{-2}$ $0$ $8.5$ $530$ $441.8$ $3.5$ $1.35$ " $0.880$ $32$ $19.8$ $2.518 \times 10^{-2}$ $0$ $8.5$ $590$ $495.0$ $3.4$ $1.35$ " $0.880$	23	18.0	$1.523 \times 10^{-1}$	0	8.5	237.	184.7	5.4	1.35	11	0.880
25 $18.4$ $1.021 \times 10^{-1}$ 0 $8.5$ $273$ $216.7$ $5.2$ $1.35$ " $0.880$ 26 $18.6$ $8.36 \times 10^{-2}$ 0 $8.5$ $305$ $241.9$ $5.0$ $1.35$ " $0.880$ 27 $18.8$ $6.843 \times 10^{-2}$ 0 $8.5$ $342$ $273.6$ $4.8$ $1.35$ " $0.880$ 28 $19.0$ $5.603 \times 10^{-2}$ 0 $8.5$ $342$ $273.6$ $4.8$ $1.35$ " $0.880$ 29 $19.2$ $4.587 \times 10^{-2}$ 0 $8.5$ $426$ $349.3$ $4.3$ $1.35$ " $0.880$ 30 $19.4$ $3.756 \times 10^{-2}$ 0 $8.5$ $475$ $393.2$ $3.9$ $1.35$ " $0.880$ 31 $19.6$ $3.075 \times 10^{-2}$ 0 $8.5$ $530$ $441.8$ $3.5$ $1.35$ " $0.880$ 32 $19.8$ $2.518 \times 10^{-2}$ 0 $8.5$ $590$ $495.0$ $3.4$ $1.35$ " $0.880$	24	18.2	$1.247 \times 10^{-1}$	0	8.5	253	198.9	5.3	1.35	FI	0.880
$26$ $18.6$ $8.36$ $x 10^{-2}$ $0$ $8.5$ $305$ $241.9$ $5.0$ $1.35$ " $0.880$ $27$ $18.8$ $6.843 \times 10^{-2}$ $0$ $8.5$ $342$ $273.6$ $4.8$ $1.35$ " $0.880$ $28$ $19.0$ $5.603 \times 10^{-2}$ $0$ $8.5$ $342$ $273.6$ $4.8$ $1.35$ " $0.880$ $29$ $19.2$ $4.587 \times 10^{-2}$ $0$ $8.5$ $426$ $349.3$ $4.3$ $1.35$ " $0.880$ $30$ $19.4$ $3.756 \times 10^{-2}$ $0$ $8.5$ $475$ $393.2$ $3.9$ $1.35$ " $0.880$ $31$ $19.6$ $3.075 \times 10^{-2}$ $0$ $8.5$ $530$ $441.8$ $3.5$ $1.35$ " $0.880$ $32$ $19.8$ $2.518 \times 10^{-2}$ $0$ $8.5$ $590$ $495.0$ $3.4$ $1.35$ " $0.880$	25	18.4	$1.021 \times 10^{-1}$	0	8.5	273	216.7	5.2	1.35	11	0.880
27 $18.8$ $6.843 \times 10^{-2}$ 0 $8.5$ $342$ $273.6$ $4.8$ $1.35$ " $0.880$ 28 $19.0$ $5.603 \times 10^{-2}$ 0 $8.5$ $386$ $311$ $4.5$ $1.35$ " $0.880$ 29 $19.2$ $4.587 \times 10^{-2}$ 0 $8.5$ $426$ $349.3$ $4.3$ $1.35$ " $0.880$ 30 $19.4$ $3.756 \times 10^{-2}$ 0 $8.5$ $475$ $393.2$ $3.9$ $1.35$ " $0.880$ 31 $19.6$ $3.075 \times 10^{-2}$ 0 $8.5$ $530$ $441.8$ $3.5$ $1.35$ " $0.880$ 32 $19.8$ $2.518 \times 10^{-2}$ 0 $8.5$ $590$ $495.0$ $3.4$ $1.35$ " $0.880$	26	<b>18.6</b>	$8.36 \times 10^{-2}$	0	8.5	305	241.9	5.0	1.35	FI	0.880
2819.0 $5.603 \times 10^{-2}$ 0 $8.5$ $386$ $311$ $4.5$ $1.35$ " $0.880$ 2919.2 $4.587 \times 10^{-2}$ 0 $8.5$ $426$ $349.3$ $4.3$ $1.35$ " $0.880$ 3019.4 $3.756 \times 10^{-2}$ 0 $8.5$ $475$ $393.2$ $3.9$ $1.35$ " $0.880$ 3119.6 $3.075 \times 10^{-2}$ 0 $8.5$ $530$ $441.8$ $3.5$ $1.35$ " $0.880$ 3219.8 $2.518 \times 10^{-2}$ 0 $8.5$ $590$ $495.0$ $3.4$ $1.35$ " $0.880$	27	18.8	$6.843 \times 10^{-2}$	0	8.5	342	273.6	4.8	1.35	11	0.880
2919.2 $4.587 \times 10^{-2}$ 08.5 $426$ $349.3$ $4.3$ $1.35$ "0.8803019.4 $3.756 \times 10^{-2}$ 08.5 $475$ $393.2$ $3.9$ $1.35$ "0.8803119.6 $3.075 \times 10^{-2}$ 08.5 $530$ $441.8$ $3.5$ 1.35"0.8803219.8 $2.518 \times 10^{-2}$ 08.5 $590$ $495.0$ $3.4$ 1.35"0.8803219.8 $2.518 \times 10^{-2}$ 08.5 $590$ $495.0$ $3.4$ 1.35"0.880	28	19.0	$5.603 \times 10^{-2}$	0	8.5	386	311	4.5	1.35	11	0.880
$30$ $19.4$ $3.756 \times 10^{-2}$ 0 $8.5$ $475$ $393.2$ $3.9$ $1.35$ " $0.880$ $31$ $19.6$ $3.075 \times 10^{-2}$ 0 $8.5$ $530$ $441.8$ $3.5$ $1.35$ " $0.880$ $32$ $19.8$ $2.518 \times 10^{-2}$ 0 $8.5$ $590$ $495.0$ $3.4$ $1.35$ " $0.880$ $32$ $19.8$ $2.518 \times 10^{-2}$ 0 $8.5$ $590$ $495.0$ $3.4$ $1.35$ " $0.880$	29	19.2	4.587 x 10 <sup>-2</sup>	0	8.5	426	349.3	4.5	1.35	Ħ	0.880
3119.6 $3.075 \times 10^{-2}$ 0 $8.5$ $530$ $441.8$ $3.5$ $1.35$ " $0.880$ 3219.8 $2.518 \times 10^{-2}$ 0 $8.5$ $590$ $495.0$ $3.4$ $1.35$ " $0.880$ 3219.8 $2.518 \times 10^{-2}$ 0 $8.5$ $590$ $495.0$ $3.4$ $1.35$ " $0.880$	30	19.4	$3.756 \times 10^{-2}$	0	8.5	475	393.2	3.9	1.35	11	0.880
32 19.8 2.518 x 10 <sup>-2</sup> 0 8.5 590 495.0 3.4 1.35 " 0.880	31	19.6	3.075 x 10-2	Ō	8.5	530	441.8	3.5	1.35	"	0.880
	32	19.8	$2.518 \times 10^{-2}$	Ō	8.5	590	495.0	3.4	1.35	n	0.880
1n 19.0 2.510 X 10 <sup>-4</sup> U I 050 55(.1 5 <sup>-7</sup> 1.55 0.215 0.880	Th	19.8	$2.518 \times 10^{-2}$	0	1	636	537.1	30	1.35	0.215	0.880

### Appendix C

SOURCE DISTRIBUTION AND CORRESPONDING CROSS-SECTION DATA FOR THE REACTOR MATERIALS

#### Appendix D

#### CALCULATIONS OF SELF-SHIELDING CORRECTIONS\*

In a system in which the flux  $\phi(x)$  is a function of a single cartesian coordinate x, diffusion theory gives the following expression for differential flux and current:

$$F(x,\mu) = \frac{\phi(x)}{2} - \frac{3D}{2} \phi'(x)\mu$$

$$G(x,\mu) = \frac{\phi(x)}{2} \mu - \frac{3D}{2} \phi'(x)\mu^{2}$$
(D.1)

Here  $\mu$  is the cosine of the angle between the neutron direction and the x axis. Integrating the terms in Eq. D.1 over all directions gives for the total flux

$$\int_{-1}^{1} F(x,\mu) d\mu = \phi(x)$$

and for the net current

$$\int_{-1}^{1} G(x,\mu) d\mu = -D\emptyset'(x) = J(x)$$

An infinite plane absorber of thickness t having absorption cross section  $\sum_{\mathbf{a}}$  is placed in the medium normal to the x axis. A neutron at  $\mathbf{x} = 0$  traveling in a direction  $\mu$  has the probability  $e^{-\sum_{\mathbf{a}} t/\mu}$  of passing through the absorber. Then

$$G(x + t,\mu) = G(x,\mu)e^{-\sum_{a}t/\mu}$$
 (D.2)

The partial current of neutrons having a component of motion in the positive x direction is

\*This approximation was derived by J. H. Marable, ORNL.

$$J_{+}(x + t) = \int_{0}^{1} G(x + t, \mu) d\mu = \int_{0}^{1} G(x, \mu) e^{-\sum_{a} t/\mu} d\mu$$
$$= \frac{\phi(x + t)}{4} - \frac{D}{2} \phi'(x + t) = \frac{\phi(x)}{4} f_{1}(\sum_{a} t) - \frac{D}{2} \phi'(x) f_{2}(\sum_{a} t) \quad (D.3)$$

where

$$f_{n}(\Sigma_{a}t) = (n+1) \int_{0}^{1} \mu^{n}e^{-\sum_{a}t/\mu} d\mu \qquad (D.4)$$

Similarly

$$j_{1}(x) = \frac{-\phi(x)}{4} - \frac{D}{2}\phi'(x) = -\frac{\phi(x+t)}{4}f_{1}(\Xi_{a}t) - \frac{D}{2}\phi'(x+t)f_{2}(\Xi_{a}t) \quad (D.5)$$

In the following, the argument of  $f_1(\sum_a t)$  and  $f_2(\sum_a t)$  are dropped for simplicity in notation . From Eqs. D.3 and D.5 the derivatives of flux are

$$\frac{D}{2}\phi'(x+t) = \frac{(1+f_1f_2)\phi(x+t)/4 - (f_1+f_2)\phi(x)/4}{(1-f_2^2)}$$
$$\frac{D}{2}\phi'(x) = \frac{-(1+f_1f_2)\phi(x)/4 + (f_1f_2)\phi(x+t)/4}{(1-f_2^2)}$$

The absorption rate per unit area in the element dp at depth p in the absorber is  $\phi(x + p)\sum_{a} dp$ . The absorption rate of neutrons entering the absorber from the left is

$$N_{\mathbf{x}} = \int_{0}^{t} \phi_{\mathbf{x}}(\mathbf{x} + \mathbf{p}) \Sigma_{\mathbf{a}} d\mathbf{p} = \Sigma_{\mathbf{a}} \int_{0}^{t} d\mathbf{p} \int_{0}^{1} \mathbf{F}(\mathbf{x} + \mathbf{p}, \mu) d\mu = \Sigma_{\mathbf{a}} \int_{0}^{t} \int_{0}^{1} \mathbf{F}(\mathbf{x}, \mu) e^{-\sum_{\mathbf{p}} / \mu} d\mu d\mathbf{p}$$

$$= \sum_{a} \int_{0}^{1} \phi(x,\mu) \frac{\mu}{Z_{a}} (1 - e^{-\sum_{a} t/\mu}) d\mu = \int_{0}^{1} \left[ \frac{\phi(x)}{2} - \frac{3D}{2} \phi'(x) \mu \right] (1 - e^{-\sum_{a} t/\mu}) \mu d\mu$$
$$= \frac{\phi(x)}{\mu} (1 - f_{1}) - \frac{p\phi'(x)}{2} (1 - f_{2}) (1 - f_{2})$$
(D.7a)

Similarly, the absorption rate of neutrons entering from the right is

$$N_{x+t} = \frac{\phi(x+t)}{4} (1 - f_1) + \frac{D\phi(x+t)}{2} (1 - f_2)$$
 (D.7b)

The total absorption rate is the sum,

$$\mathbb{N} = \frac{(1 - f_1)}{2} \frac{\phi(x) + \phi(x + t)}{2} - (1 - f_2) \frac{D}{2} \left[ \phi'(x) - \phi'(x + t) \right] \quad (D.7c)$$

Substituting the expressions from Eq. D.6 into D.7c gives the absorption rate

$$N = \left(\frac{1 - f_1}{1 + f_2}\right) \left(\frac{\phi(x) + \phi(x + t)}{2}\right) = \left(\frac{1 - f_1}{1 + f_2}\right) \overline{\phi}$$
(D.8)

where  $\overline{\emptyset}$  is the average of the two boundary values of the flux. Assuming no self shielding, the absorption rate would be  $N_0 = \overline{\emptyset} \sum_a t$ . The self-shielding factor is therefore the ratio

$$S_{1} = \frac{1}{\lambda_{a}^{\prime} t \overline{\emptyset}} \left( \frac{1 - f_{1}}{1 + f_{2}} \right) \overline{\emptyset} = \frac{1}{\lambda_{a}^{\prime} t} \left( \frac{1 - f_{1}}{1 + f_{2}} \right)$$
(D.9)

This is the  $p_1$  self-shielding factor and may be compared to  $p_0$  approximation which has the form

$$S_{o} = \frac{1}{\sum_{a} t} \left( \frac{1 - f_{1}}{2} \right)$$
(D.10)

The functions  $S_1$  and  $S_2$  are shown in Fig. 14, p. 27 .

A similar approach has been used to calculate the activation rate of a foil covered on both sides by similar absorbers. Using the notation  $x = \sum_{a} t$  with subscript c for the covers and d for the detector foil, the activation rate of the detector per unit area per unit time is

$$N = \overline{\phi} \frac{f_1(x_c) - f_1(x_c + x_o)}{2} + \frac{\left[1 - f_1(x_d + 2x_c)\right] \left[f_2(x_c) - f_2(x_d + x_c)\right]}{2 \left[1 + f_2(x_d + 2x_c)\right]}$$
(D.11)  
Consider the function  $f_n(x) \equiv (n + 1) \int_0^1 \mu^n e^{-x/\mu} d\mu$  given in Eq. D.4.

Direct partial integration gives the recursion relation

$$f_n(x) = -\frac{x}{n} f_{n-1}(x) + e^{-x}$$
 (D.12)

Also note that<sup>1</sup>

$$f_{0}(x) = \int_{0}^{1} e^{-x/\mu} d\mu = e^{-x} + xEi(-x)$$
 (D.13)

By Eqs. D.12 and D.13

$$f_{0}(x) = e^{-x} + xEi(-x)$$

$$f_{1}(x) = (1 - x)e^{-x} - x^{2}Ei(-x)$$

$$f_{2}(x) = (1 - \frac{x(1 - x)}{2})e^{-x} + \frac{x^{2}}{2}Ei(-x)$$
(D.14)

1. The function of  $-\text{Ei}(-x) \equiv \int_{x}^{\infty} \frac{e^{-t}}{t} dt$  is the exponential integral, the values of which are tabulated. See, for example, "Table of Functions," by Jahnke and Ende, Dover Publications (1945).

The function f(x) is related to the tabulated function<sup>2</sup>

(

$$\mathbf{E}_{\mathbf{n}}(\mathbf{x}) \equiv \int_{1}^{\infty} e^{-\mathbf{X}\mathbf{\mu}} \mathbf{\mu}^{-\mathbf{n}} d\mathbf{\mu}$$

ЪУ

$$E_n(x) = \frac{f_{n-2}(x)}{n-1}$$
 (D.15)

3

or

•

$$f_n(x) = (n + 1)E_{n+2}(x).$$

2. G. Placzek, "The Functions 
$$E_n(x) \equiv \int_{1}^{\infty} e^{-x\mu} \mu^{-n} d\mu$$
," MT-1.

#### Appendix E

### CALCULATIONS OF $\not\in$ FOR ANISOTROPIC SCATTERING

Consider a neutron of energy E, which is elastically scattered by a nucleus of atomic mass A. Defining  $\mu$  as the cosine of the scattering angle in the center of mass system, the energy,  $E_2$ , of the neutron after collision is given by

$$E_{2} = E_{1}\left[\left(\frac{1+\alpha}{2}\right) + \left(\frac{1-\alpha}{2}\right)\mu\right]$$
(E.1)

where  $\alpha \equiv \left(\frac{A-1}{A+1}\right)^2$ .

In isotropic scattering the frequency function for scattering into a range du about  $\mu$  is given by

$$f_{o}(\mu) = \frac{1}{2}$$
, which satisfies the normalization  $\int_{-1}^{1} f_{o}(\mu)d\mu = 1.$  (E.2)

The average  $\mu$  is zero in this case.

In anisotropic scattering one has a frequency function

$$f(\mu) = f_{0}(\mu) + f_{1}(\mu) = \frac{1}{2} + f_{1}(\mu).$$
Normalization requires that 
$$\int_{-1}^{1} f_{1}(\mu)d\mu = 0$$
(E.3)

The average cosine,  $\bar{\mu},$  may be measured experimentally and is given by

$$\overline{\mu} = \int_{-1}^{1} \mu f(\mu) d\mu = \int_{-1}^{1} \mu f_{1}(\mu) d\mu. \quad (E.4)$$

The average lethargy gain per collision,  $\xi$ , may be calculated immediately.

$$\underbrace{ \left\{ = \frac{1}{1 + \frac{E_1}{E_2}} = \int_{-1}^{1} \left( \ln \frac{E_1}{E_2} \right) \mathbf{f}(\mu) d\mu = \int_{-1}^{1} \ln \frac{E_1}{E_2} \left[ \mathbf{f}_0(\mu) + \mathbf{f}_1(\mu) \right] d\mu$$

$$= -\int_{-1}^{1} \ln\left(\frac{1+\alpha}{2} + \frac{1-\alpha}{2}\mu\right) \left(\frac{1}{2}\right) d\mu - \int_{-1}^{1} \ln\left(\frac{1+\alpha}{2} + \frac{1-\alpha}{2}\mu\right) f_{1}(\mu) d\mu$$

$$= \underbrace{z}_{\neq 0} - \int_{-1}^{1} \left[ \ln \left( \frac{1+\alpha}{2} \right) + \ln \left( 1 + \frac{1-\alpha}{1+\alpha} \mu \right) \right] f_{1}(\mu) d\mu$$
$$= \underbrace{z}_{\neq 0} - \ln \left( \frac{1+\alpha}{2} \right) \int_{-1}^{1} f_{1}(\mu) d\mu - \int_{-1}^{1} \left[ \frac{1-\alpha}{1+\alpha} \mu - \frac{1}{2} \left( \frac{1-\alpha}{1+\alpha} \right)^{2} \mu + \cdots \right] f_{1}(\mu) d\mu$$

$$= \xi_{0} - \frac{1-\alpha}{1+\alpha} \int_{-1}^{1} \mu f_{1}(\mu) d\mu + \frac{1}{2} \left(\frac{1-\alpha}{1+\alpha}\right)^{2} \int_{-1}^{1} \mu^{2} f_{1}(\mu) d\mu + \cdots$$

and finally

$$\xi \stackrel{\sim}{=} \xi_0 - \frac{1-\alpha}{1+\alpha} \overline{\mu} , \qquad (E.5)$$

using Eqs. E.3 and E.4 and ignoring higher order terms in  $\left(\frac{1-\alpha}{1+\alpha}\right)$ ,  $\xi_0$  is the value of  $\xi$  for the isotropic scattering case. The approximation is convenient if  $\mu$  is known experimentally. Ignoring higher order terms in  $\left(\frac{1-\alpha}{1+\alpha}\right)$  is not serious for heavy elements, e.g., in the case of beryllium where A = 9,  $\frac{1}{2}\left(\frac{1-\alpha}{1+\alpha}\right)^2 = 0.024$  compared to  $\left(\frac{1-\alpha}{1+\alpha}\right) = 0.22$ . Using the value of  $\mu$  from page 28 the value of  $\xi$  is found to be 0.206 - (0.220)(0.254) = 0.150.

#### Appendix F

#### CORRECTION FOR EXTERNAL ALUMINUM GRID

Consider a spherical reactor having a core radius r and an outside reflector radius R. The reflector is nonabsorbing material having a scattering mean free path

$$\lambda_{\rm s} = \frac{1}{\Sigma_{\rm s}} > {\rm R} - {\rm r}.$$

A fraction of the neutrons leaking out of the core will undergo collisions in the reflector and may eventually be returned to the core. Only first collisions are considered in this approximation. From Fig. F-la

$$\rho = r \left[ \frac{R^2 - r^2}{r^2} + \mu^2 \right]^{1/2} - \mu$$
 (F.1)

where  $\mu = \cos \theta$ .

Assuming the leakage to be constant in  $\mu$ , the average of  $\rho$  is found to be

$$\overline{e} = \frac{\int_{0}^{1} e^{d\mu}}{\int_{0}^{1} d\mu} = \frac{R - r}{2} + \frac{R^{2} - r^{2}}{4r} \ln\left(\frac{R + r}{R - r}\right) \quad (F.2)$$

If the total leakage rate from the core is E, the number of first collisions per unit time in the reflector will be  $n = E \Sigma_s \overline{\rho}$ . (F.3)

Assuming the first-collision density to have an inverse square distribution, the first-collision density in a volume element  $x^2d$ -Adx (see Fig. F-lb), is

$$\frac{ndxd \Omega}{4\pi(R-r)}$$
(F.4)



Fig. F-1. Spherical Reactor Geometry for Calculation of Leakage Correction Which Includes Effect of Aluminum Grid.

The fraction of solid angle subtended by the core at a distance x is

$$\frac{1-\cos \frac{\gamma}{2}}{2} = \frac{1}{2} \left[ 1 - \frac{(x^2 - r^2)^{1/2}}{x} \right].$$
 (F.5)

Assuming isotropic scattering and combining Eqs. F.3, F.4, and F.5 the rate of return to the core is

$$\mathbf{F} = \int_{\mathbf{x}=\mathbf{r}}^{\mathbf{R}} \int_{\Omega} \left(\frac{1}{2}\right) \left(1 - \frac{(\mathbf{x}^2 - \mathbf{r}^2)^{1/2}}{\mathbf{x}}\right) \left(\frac{\mathrm{nd}\mathbf{x}\mathrm{d}\Omega}{4\pi(\mathbf{R} - \mathbf{r})}\right)$$
$$= \frac{\mathrm{E}\Sigma_{\mathbf{s}}}{4} \left(1 + \frac{\mathbf{R} + \mathbf{r}}{2\mathbf{r}} \ln \frac{\mathbf{R} + \mathbf{r}}{\mathbf{R} - \mathbf{r}}\right) \left[(\mathbf{R} - \mathbf{r}) - (\mathbf{R}^2 - \mathbf{r}^2)^{1/2} + \mathbf{r} \cos^{-1} \frac{\mathbf{r}}{\mathbf{R}}\right] \quad (F.6)$$

and the fraction of leakage neutrons which are returned is

$$f = \frac{F}{E} = \frac{\Sigma_{s}}{4} \left[ 1 + \frac{R+r}{2r} \ln \frac{R+r}{R-r} \right] \left[ (R-r) - (R^{2} - r^{2})^{1/2} + r \cos^{-1} \frac{r}{R} \right]$$
(F.7)

Neutrons which reenter the core are assumed to be distributed uniformly throughout the core and are less effective than the same number of neutrons would be in a normal distribution.

In a parallelpiped bare reactor of dimension  $2a \ge 2b \ge 2c$ , the flux is given by

$$\phi(\mathbf{x},\mathbf{y},\mathbf{z}) = \phi_0 \cos \frac{\mathbf{x}\pi}{2\mathbf{a}} \cos \frac{\mathbf{y}\pi}{2\mathbf{b}} \cos \frac{\mathbf{z}\pi}{2\mathbf{c}} . \qquad (F.8)$$

In the normal mode the source density is given by

$$q(x,y,z) = q_0 \left( \cos \frac{x\pi}{2a} \right) \left( \cos \frac{y\pi}{2b} \right) \left( \cos \frac{z\pi}{2c} \right),$$
 (F.9)

where the total source, Q is  $q_0 \frac{8V}{\pi^3}$ , and V is the volume of the reactor. In a uniform distribution the same number of neutrons would give rise to a source density  $\overline{q} = \frac{Q}{V} = \frac{8}{\pi^3} q_0$ .

Assuming the importance function for a source to be proportional to the flux, the relative effectiveness of a uniform distribution compared to that for a normal mode distribution is:

0

$$\frac{\text{I uniform}}{\text{I normal}} = \frac{\sqrt[Vol]{q\phi(x,y,z)dv}}{\int \overline{q}(x,y,z)\phi(x,y,z)dv} = \left(\frac{8}{\pi^2}\right)^3 = 0.53256 \quad (F.10)$$

Comparing CA-1 and CA-18 to a volume equivalent spherical system, one finds from Eqs. F.7 and F.10 the value of the effective f, i.e., the leakage correction discussed in Eq. F.7:

	<u>CA-1</u>	<u>CA-18</u>
r	34.22 cm	40.60 cm
R	134.5 cm	134.5 cm
f .	8.646 Σ <sub>s</sub>	9 <b>.</b> 960 Σ <sub>ε</sub>
Effective f	4.605 Σ <sub>s</sub>	5 <b>.3</b> 04 Σ <sub>s</sub>

### Appendix G

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